

# Multiobjective Optimisation of Hybrid Wind-PV-Battery-Fuel Cell-Electrolyser-Diesel Systems: An Integrated Configuration-Size Formulation Approach

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## Abstract

A generic integrated configuration-size optimisation formulation for design of hybrid renewable energy systems (HRES) is presented in this paper. This formulation allows identifying the optimum configuration for a given site and the optimum size of each component in that configuration by solving only one optimisation problem. Single and multiobjective case studies are defined for both on-grid and standalone systems using wind turbine, PV panel, battery bank, fuel cell, electrolyser and diesel generator as potential components. To solve the optimisation problems a genetic algorithm (GA) and a nondominated sorting GA (NSGA-II) are developed, in which the reproduction operators are designed carefully for robust exploration and exploitation at both size and configuration levels. Eight single and multiobjective case studies for a variety of renewable resources, objectives and constraints are conducted. The results show the versatility of the problem formulation in defining different HRES design problems and the robustness of the developed GA and NSGA-II in search within the design space at both configuration and size levels and finding the optimum size and configuration simultaneously.

Keywords: integrated design; hybrid renewable energy systems; GA; NSGA-II; optimisation; MOHRES

## 1 Introduction

Size optimisation of hybrid renewable energy systems (HRES) is one of the branches of research on HRES, which has received significant attention from the research community. A review of the recently published literature in this field shows that researchers have taken different approaches and methods in (i) problem formulation, (ii) HRES analysis, and (iii) optimisation method.

Problem formulation depends on the type and the number of design variables, objectives and constraints included in the optimisation problem. In the context of size optimisation of HRES, design variables are the size of components in the configuration. Almost all technologically feasible configurations have been the subject of recent studies for power and combined heat and power generation. While the majority of the publications have reported a single objective optimisation, there is an increasing trend for adopting multiobjective optimisation approach in more recent publications. Optimisation method adapted for size optimisation of HRES has been under the influence of the analysis method and the problem formulation. Genetic Algorithm (GA) and Particle Swarm (PS), and their multiobjective variations seem to be the most popular metaheuristic methods for this field.

51

52 Analysis of HRES varies from one study to another, depending on the methodological  
53 approach in system characterisation, the level of details of cost and power models, and the  
54 performance measures of interest. In terms of handling the uncertainties in the demand load,  
55 renewable resources, and models, the reported studies have taken two approaches, namely,  
56 deterministic or nondeterministic. While the majority of the published works are based on  
57 deterministic approaches, some researchers have adopted a nondeterministic approach more  
58 recently.

59

60 Atieh et al [1] included PV, battery bank, and diesel generator in their study. They used a GA  
61 to find the optimum size of these components by considering the total lifespan cost (TLSC)  
62 as the objective. Ismail et al [2] also used GA to solve their PV-Battery-Diesel size  
63 optimisation problem, but with levelised cost of energy (LCE) as the optimisation objective.  
64 Samy et al [3] conducted size optimisation for PV-Fuel cell configuration using PS, Flower  
65 Pollination, and Artificial Bee Colony as their single-objective optimisation methods. They  
66 used total annual cost (TAC) and loss of power supply probability (LPSP) as objectives.  
67 Senthil Kumar et al [4] also reported a size optimisation of PV-Fuel cell configuration by  
68 using Nelder-Mead algorithm and a hybrid method as their optimisation algorithm.  
69 Mahmoudimehr and Shabani conducted a multiobjective size optimisation for a PV- Pumped  
70 hydro configuration [5]. They considered net present value (NPV) and LPSP as the  
71 optimisation objectives and used a Nondominated Sorting Genetic Algorithm (NSGA) as the  
72 search method. Mohamed et al [6] used a GA with TAC as the objective to solve the size  
73 optimisation of a PV-Wind configuration. PV-Wind-Battery is one of the most studied  
74 configurations [7-16]. Acuna et al [7] conducted a multiobjective size optimisation, with  
75 NPV, LPSP, and LCE as objectives. They used NSGA as their optimisation method. Ahmadi  
76 and Abdi conducted a single objective size optimisation of PV-Wind-Battery system with  
77 LCE as the objective using big-bang big- crunch method [8]. Fetanatand Khorasaninejad [9]  
78 used an Ant Colony algorithm for size optimization of PV-Wind-Battery configuration in a  
79 continuous domain with the objective of LCE. They used a continuous domain. Another  
80 example of multiobjective size optimisation of PV-Wind-Battery systems with NPV, LPSP,  
81 and LCE as objectives, is reported in Ghorbani et al [10]. They used multiobjective PS to  
82 solve the problem. [11] conducted single objective size optimisation using the tool DER-  
83 CAM for LCE and CO2 emission. Kamjoo et al [12] used deficiency of power supply  
84 probability (DPSP) and system cost as two objectives of their multiobjective size  
85 optimisation. The works reported in [13, 14, 15, and 16] are other examples of single  
86 objective size optimisation of PV-Wind-Battery configuration, in which LCE is used as the  
87 optimisation objective. The optimisation methods, however, are different: Bee Swarm  
88 algorithm [13], built-in optimiser in HOMER [14], PS and Harmony Search algorithms [15],  
89 and a hybrid GA-exhaustive search method [16]. PV-Wind-Battery-Diesel is another well-  
90 studied configuration, mainly as a standalone HRES [17-25]. In [17], a hybrid Harmony  
91 Search-Simulated Annealing optimisation algorithm is used to minimise the LCE of the  
92 system with bio-diesel. TLSC is the optimisation objective for PV-Wind-Battery-Diesel size  
93 optimisations reported in [18, 19, and 20] using different optimisation methods: response  
94 surface [18], PS [19], and GA [20]. Ogunjuyigbe et al [21] used a single objective GA to find  
95 the optimal size of a PV-Wind-Battery-Diesel system which satisfies a set of constraints on  
96 TLSC, CO2 emission, and the excess power. LCE and renewable penetration are the  
97 objectives of the PV-Wind-Battery-Diesel multiobjective size optimisations reported in [22].  
98 References [23, 24, and 25] also report the results of multiobjective size optimisation of PV-  
99 Wind-Battery-Diesel configuration with LCE and LPSP [23]; LCE, NPV, and LPSP [24]; and  
100 LCE, LPSP, and renewable penetration [25] as objectives. Maheri [26], Sharafi and

101 ELMekrawy [27], and Maleki and Pourfayaz [28] considered PV-Wind-Battery-Diesel-Fuel  
102 cell configuration in their studies. In [26] using a GA with the objectives of TLSC and unmet  
103 load, the effect of dispatch strategy on the performance of hybrid wind-PV-battery-diesel-fuel  
104 cell systems was studied. In [27], a combination of TLSC, unmet load, and emission serves as  
105 an aggregate objective function for the PS algorithm to find the optimum size of each  
106 component in the HRES. Maleki and Pourfayaz used LCE as the objective and used a  
107 harmony search algorithm in their study reported in [28]. Other system configurations which  
108 have been studied include PV-Wind-Battery- Fuel cell [29], PV-Wind-Biomass [30], PV-  
109 Wind-Biomass-Biodiesel [31], PV-Wind-Battery-Thermal storage [32], PV-Wind-Diesel  
110 [33], PV-Wind-Fuel cell [34], PV-Wind-Pumped hydro [35], PV-Wind-Solar heat collector-  
111 Biomass-Heat pump-Thermal storage [36], Wind-Battery-Diesel [37], Wind-Fuel cell [38],  
112 PV-Wind-Electric vehicle [39], Wind-PV- Thermal storage tank [40], and PV-Hydrogen  
113 storage-Fuel cell -Solar Thermal [41]. Besides power and combined heat and power  
114 generation, size optimisation of renewable and storage systems can be defined in the context  
115 of energy transition and system integration scenarios, for instance, renewable-based heating  
116 sector [42], cross-sectoral integration of HRES [43], smart energy integration [44], and  
117 energy planning [45]. Besides the cost and reliability-rated objectives, CO<sub>2</sub> emission and  
118 renewable penetration are commonly used objectives in size optimisation of hybrid  
119 renewable-nonrenewable systems [46-49].

120

121 In all reported research dealing with the optimisation of HRES, the configuration of the  
122 system is decided prior to the size optimisation. In other words, design of a HRES takes place  
123 in two sequential stages. First the configuration of the system and the type of renewable, non-  
124 renewable and storage/auxiliary components are decided. In the second stage, the optimum  
125 size of each component in the system is found. A sequential design optimisation has a major  
126 drawback. In many cases the decision on the configuration cannot be made unless a detailed  
127 cost-benefit analysis is carried out. In such cases, for each possible configuration a size  
128 optimisation problem must be solved to determine the optimal cost-related and reliability-  
129 related performance measures for that configuration. Once the optimal design of all possible  
130 configurations is done, the best configuration can be identified. This, however, can be a  
131 cumbersome task. For instance, there are 21 possible configurations with at least one  
132 renewable and one auxiliary/storage component which can be made with wind turbine, PV  
133 panel, battery bank, fuel cell/electrolyser and diesel generator as potential components.

134

135 On the other hand, an integrated configuration-size optimisation formulation has a clear  
136 advantage to the current practice, in which the size optimisation is conducted for a pre-  
137 defined and not-necessarily optimum configuration or all potential configurations. This paper  
138 presents an integrated configuration-size optimisation formulation, which allows finding the  
139 optimum configuration for a given site and application, and the optimum size of each  
140 component in that configuration by solving only one optimisation problem.

141

142 All the optimisation methods reported in the literature are designed and tested for size  
143 optimisation problems only and therefore are not robust enough for solving an integrated  
144 configuration-size optimisation problem. In an integrated configuration-size optimisation  
145 problem the search in the design space must take place at both configuration and size levels.  
146 Novel genetic reproduction operators which are designed specifically for simultaneous  
147 exploration and exploitation at both levels are also developed and presented in this paper.

148

## 149 2 Integrated Configuration-Size Formulation

150 Following common practice in formulating an optimisation problem, this section starts by  
151 defining the design variables in Section 2.1 and then in Section 2.2 elaborates on the design  
152 space, or in other words, the realistic bounds for design variables in which we search for  
153 optimum solution(s). The basis of the proposed integrated configuration-size optimisation  
154 formulation is explained in this section. In Section 2.3 the design qualities of the problem at  
155 hand, here, the performance measure of hybrid renewable energy systems, are defined. The  
156 system model, correlating the design variables to the design qualities are also given in this  
157 section after defining each performance measure. Finally, in Section 2.4 the optimisation  
158 problem is formulated in standard format and is reported.

### 160 2.1 Design Variables

161 In integrated configuration-size optimisation, starting from a generic configuration (in this  
162 study wind-PV-fuel cell-electrolyser-diesel-battery bank configuration), the vector of design  
163 variables is defined as:

$$165 \vec{X} = \{n_{WT}, R_{WT}, A_{PV}, n_B, P_{D,nom}, P_{FC,nom}, P_{EL,nom}\} \quad (1)$$

166 where,  $n_{WT}$  stands for the number of wind turbines,  $R_{WT}$  is the wind turbine rotor radius  
167 (representing the size of the wind turbine),  $A_{PV}$  is the total size of the PV panels,  $n_B$  stands  
168 for the number of batteries, and  $P_{D,nom}$ ,  $P_{FC,nom}$  and  $P_{EL,nom}$ , respectively, are the nominal  
169 power of the diesel generator, fuel cell and electrolyser.

170 It should be noted that, as explained below, there are cases that one of the design variables  
171 associated to wind energy ( $n_{WT}, R_{WT}$ ) is excluded from the vector of design variables. If the  
172 wind turbine to be used in the system is known, then  $R_{WT}$  is known and fixed, leaving  $n_{WT} \geq$   
173 0 as the only design variable to be decided. If the wind turbine has not been decided prior to  
174 the optimisation, and if the optimum amount of wind energy can be delivered by one wind  
175 turbine, then  $n_{WT} = 1$  and fixed, leaving  $R_{WT}$  as the remaining design variable to be found. It  
176 is a well-known fact that the levelised cost of wind energy reduces with the size of wind  
177 turbine. That is, one wind turbine with a nominal power of  $P$  is cheaper than  $n$  wind turbines,  
178 each with a nominal power of  $P/n$ . If the wind turbine has not been decided prior to the  
179 optimisation, and if the optimum amount of wind energy cannot be delivered by one wind  
180 turbine, then both  $n_{WT}$  and  $R_{WT}$  are design variables.

### 184 2.2 Search Space

185 These design variables are assumed to be bounded between the following ultimate lower and  
186 upper limits:

$$188 \vec{X}^l = \{n_{WT}^l, 0, 0, 0, 0, 0, 0\} \quad (2.a)$$

189 where, as explained above:

$$192 n_{WT}^l = \begin{cases} 0 & \text{if wind turbine is known} \\ 1 & \text{if wind turbine is not known} \end{cases} \quad (2.b)$$

$$194 \vec{X}^u = \{n_{WT}^u, R_{WT}^u, A_{PV}^u, n_B^u, P_{D,nom}^u, P_{FC,nom}^u, P_{EL,nom}^u\} \quad (3)$$

195

196 Using the ultimate upper and lower limits as given by Equations 2 and 3, the optimisation  
 197 process explores a generic configuration including all components. However, within the  
 198 optimisation process, the size of components which cannot produce/store energy efficiently  
 199 approach zero, leading to their exclusion from the configuration. In other words, the  
 200 configuration is obtained via a full cost/performance analysis and an integrated configuration-  
 201 size optimisation rather than just being selected by the designer.  
 202

203 The upper limits for design variables  $\vec{X}^u$  must be selected carefully. On one hand, we want to  
 204 ensure that the search space is large enough to include all potential optimum solutions, while  
 205 on the other hand we want to increase the robustness of the search process by avoiding search  
 206 in unrealistic/infeasible part of the search domain.  
 207

208 In this study, the upper limit for the wind turbine rotor radius  $R_{WT}^u$  and the upper limit for the  
 209 number of wind turbines  $n_{WT}^u$  are calculated based on the assumption that the wind turbine(s),  
 210 by its own, can produce enough power to supply the demand load. This assumption leads to:  
 211

$$212 \quad R_{WT}^u = \sqrt{\frac{L_{max}}{0.5\rho\pi V_{min}^3 C_p \eta_{EM}}} \quad (4.a)$$

213  
 214 and

$$215 \quad n_{WT}^u = \left\lceil \frac{L_{max}}{0.5\rho\pi (R_{WT}^u)^2 V_{min}^3 C_p \eta_{EM}} \right\rceil \quad (4.b)$$

217 where,  $\rho$  is the air density,  $C_p$  stands for the power coefficient,  $\eta_{EM}$  is electrical/mechanical  
 218 overall efficiency,  $L_{max}$  is the maximum hourly-averaged peak demand throughout the year  
 219 and  $V_{min}$  is the minimum hourly-averaged wind speed throughout the whole year at a  
 220 conservatively low hub elevation (in this study,  $h_{hub,0} = 12 \text{ m}$ ) calculated using the  
 221 logarithmic law.  
 222  
 223

224 In Equation 4.a the term  $0.5\rho V_{min}^3 C_p \eta_{EM}$  is the wind turbine power per unit rotor area,  
 225  $\pi R_{WT}^2$ . It should be noted that  $R_{WT}^u$  obtained from Equation 4.a must not exceed the rotor  
 226 radius of the largest available wind turbine. For instance, if we assume Vestas V164-8.0 is the  
 227 largest available wind turbine, then  $R_{WT}^u$  is limited to 82m. In Equation 4.b,  $\lceil \dots \rceil$  stands for  
 228 rounding up to the nearest integer number. The denominator of the right-hand side of  
 229 Equation 4.b is the maximum power that can be produced by the largest wind turbine  
 230 available.  
 231

232 Since the rotor power coefficient depends on the wind speed, very conservative (low) value  
 233 for the power coefficient is also assumed ( $C_p = 0.2$ ). A conservative value for the combined  
 234 efficiency of the electrical components and the gear train is also used  $\eta_{EM} = 80\%$ .  
 235

236 The upper limit for the PV panel area can be calculated by taking the same approach but  
 237 based on daily-average data rather than hourly-averaged data. This is because the peak load  
 238 might happen in hours that there is no sunlight and in practice PV panels always are  
 239 accompanied by a storage component for standalone systems. The upper limit for the PV  
 240 area,  $A_{PV}^u$ , can be found by:  
 241

$$A_{PV}^u = \frac{L_{d,max}}{I_{d,min}\eta_{PV}} \quad (5)$$

243

244 where,  $L_{d,max}$  is the maximum daily-averaged demand throughout the year,  $I_{d,min}$  is the  
 245 minimum daily-averaged solar irradiance throughout the whole year, and  $\eta_{PV}$  is a  
 246 conservative average value for the PV panel efficiency (for example 10%). The denominator  
 247 of the right-hand side of Equation 5 is the PV power per unit area.

248

249 The upper size limit for the nominal power of the diesel generator is obtained based on the  
 250 maximum hourly-averaged peak demand throughout the year  $L_{max}$  as follows:

251

$$P_{D,nom}^u = \frac{L_{max}(1+MOS)}{\eta_D} \quad (6)$$

253

254 where,  $\eta_D = 0.4$ , a conservative approximation for the diesel efficiency, and  $(1 + MOS)$  is  
 255 the load factor based on a margin of safety ( $MOS$ ).

256

257 Adopting the same approach for the fuel cell, assuming a conservative approximate  
 258 efficiency of  $\eta_{FC} = 0.47$ , the upper size limit for the nominal power of the fuel cell is  
 259 obtained

260

$$P_{FC,nom}^u = \frac{L_{max}((1+MOS))}{\eta_{FC}} \quad (7)$$

262

263 The capacity of the electrolyser should be large enough to produce enough hydrogen for the  
 264 fuel cell operation within one hour. Hence, the upper limit of the electrolyser size  $P_{EL,nom}^u$  is  
 265 associated to the upper limit of the fuel cell size  $P_{FC,nom}^u$ :

266

$$P_{EL,nom}^u = P_{FC,nom}^u / \eta_{EL} \quad (8)$$

268

269 where, the efficiency of the electrolyser in this study is taken as  $\eta_{EL} = 0.74$ .

270

271 The upper limit for the number of batteries in the battery bank,  $n_B^u$  is calculated based on the  
 272 assumption that the battery bank can store enough energy to supply the demand load for an  
 273 autonomy period  $T_{a,B}$ , normally taken as 1 day. Therefore, using the maximum daily-  
 274 averaged demand load through the whole year,  $L_{d,max}$ , the upper limit for the number of  
 275 batteries can be determined by:

276

$$n_B^u = \frac{T_{a,B}L_{d,max}(1+MOS)}{(1-SOC_{min})c_B V_B \eta_{B,d}} \quad (9)$$

278

279 where,  $\eta_{B,d}$  is the battery efficiency in discharge, and  $SOC_{min}$  is the permissible minimum  
 280 SOC without causing damage to the batteries, and  $1 - SOC_{min}$  is the proportion of the battery  
 281 bank capacity that can be used. In Equation 9, the nominator and denominator of the right-  
 282 hand side are the required power to supply demand load for the autonomy period of  $T_{a,B}$  with  
 283 a margin of safety  $MoS$  and the extractable energy stored in a single fully charged battery in  
 284 the battery bank respectively.

285

286 The general integrated configuration-size optimisation formulation above allows for  
 287 formulation of special cases as well. There are cases that the designer wants to design a

288 specific configuration or fix the size of component prior to the optimisation. To do these the  
 289 designer needs to use alternative upper and lower limits. By setting both the lower and upper  
 290 limits of a component equal to zero, that component is excluded from the configuration. By  
 291 doing so, the designer can fix the configuration. The formulation above also allows for pre-  
 292 sizing of components. Pre-sizing refers to the selection of the size of a component outside the  
 293 optimisation process. In order to exclude the size of a component from the set of design  
 294 variables but keep the component in the HRES configuration the upper and lower limits are  
 295 identical and set as the size of that component.

296

### 297 **2.3 Design Qualities (Performance Measures)**

298 In this study the following design qualities or performance measures are used in evaluating a  
 299 HRES.

300

301 **Unmet Load**-Unmet load, the difference between the available power and the demand load,  
 302 is the part of the demand load that is not supplied by the HRES. Using hourly averaged data,  
 303 the unmet load  $U$  and the total annual unmet load  $U_t$  are defined as:

304

$$305 \quad U = \begin{cases} L - P_a & \text{if } L > P_a \\ 0 & \text{if } L \leq P_a \end{cases} \quad (10)$$

306

307 and

308

$$309 \quad U_t = \sum_{i=1}^{8760} U_i \quad (11)$$

310

311 where,  $L$  and  $P_a$ , respectively, are the hourly averaged demand load and available power  
 312 from the system. For a system with wind, PV, battery, fuel cell and diesel,  $P_a$  is given by:

313

$$314 \quad P_a = P_{WT} + P_{PV} + P_{B,e} + P_{FC,e} + P_{D,nom} \quad (12)$$

315

316 in which,  $P_{WT}$  and  $P_{PV}$  are the power produced by wind turbine(s) and PV panels,  $P_{B,e}$  and  
 317  $P_{FC,e}$  are the extractable power from the battery bank and fuel cell and  $P_{D,nom}$  is the total  
 318 nominal power of the diesel generator(s).

319

320 Wind power  $P_{WT}$  is given by:

321

$$322 \quad P_{WT} = n_{WT} \frac{1}{2} \pi \rho V_{hub}^3 R_{WT}^2 C_p \eta_{EM} \quad (13)$$

323

324 In case the wind turbine to be used in the system is known, the rotor power coefficient  $C_p$  at  
 325 various wind speeds,  $C_p(V_{hub})$ , is given in the turbine specification sheet and hence it is a  
 326 known parameter. The same applies for  $\eta_{EM}$ . The hub height  $h_{hub}$  is also a known parameter  
 327 and using either of logarithmic or power laws  $V_{hub}$  is found readily.

328

329 In case the wind turbine is not selected/known prior to the optimisation process, model given  
 330 in [20], or similar models, can be used for  $C_p(V_{hub})$ . For the overall wind turbine mechanical  
 331 and electrical efficiency  $\eta_{EM}$ , a reasonable value (e.g. 90%) is used. In this case, the height of  
 332 the hub depends on the size of the turbine, which is unknown at the start of the design phase.

333 The following rule of thumb is used to estimate the hub height:

334

$$335 \quad h_{hub} = \max\{h_c + R_{WT}, 2R_{WT}\} \quad (14)$$

336  
 337 where,  $h_c$  is the minimum ground clearance for the blade tip.  
 338

339 Solar PV power  $P_{PV}$  is given by:

$$340 \quad P_{PV} = IA_{PV}\eta_{PV} \quad (15)$$

341 where,  $I$  is the hourly averaged solar irradiance in  $W/m^2$ ,  $A_{PV}$  is the total area of the solar  
 342 panels, and  $\eta_{PV}$  is the overall efficiency of the of the PV panels.  
 343

344 The extractable power from the battery bank  $P_{B,e}$  is given by:

$$345 \quad P_{B,e} = (SOC - SOC_{min})n_B c_B V_B \eta_B \quad (16)$$

346 where,  $c_B$  (Ah) and  $\eta_B$  stand for the unit nominal capacity and efficiency of batteries in  
 347 discharge;  $n_B$  is the number of batteries in the bank;  $SOC$  is the state of charge of the battery  
 348 bank, and  $SOC_{min}$  is the permissible minimum state of charge of the batter.  
 349

350 The extractable power from the hydrogen tank through a fuel cell, depends on the extractable  
 351 mass of the hydrogen from the tank  $M_{H_2,e}$  and the fuel cell efficiency  $\eta_{FC}$  and it is limited to  
 352 the nominal power of fuel cell:  
 353

$$354 \quad P_{FC,e} = \min\{P_{FC,nom}, M_{H_2,e} m_{H_2} LHV \eta_{FC}\} \quad (17)$$

355 where,  $m_{H_2} = 2.016 \times 10^{-3} \text{ kg/mol}$  and  $LHV = 33000 \text{ Wh/kg}$  are the molar mass and  
 356 lower heating value of hydrogen,  $P_{FC,nom}$  is the fuel cell nominal power and  $M_{H_2,e}$ , the  
 357 extractable mass of hydrogen from hydrogen tank is the difference between the mass of  
 358 stored hydrogen in the tank,  $M_{H_2}$  and  $M_{H_2,min}$ , the unextractable mass from the tank due to  
 359 drop in the tank pressure:  
 360

$$361 \quad M_{H_2,e} = M_{H_2} - M_{H_2,min} \quad (18)$$

362 **Feed-in and Profit-In** standalone systems the renewable power primarily supplies the  
 363 demand load and, where applicable, if there is any excess power it is used to charge the  
 364 battery bank and the hydrogen tank. Any excess power beyond this will be dumped. In grid  
 365 connected systems, this excess power can be sold to the grid instead. The excess power that  
 366 can be feed-in to the grid is given by:  
 367

$$368 \quad P_{ex} = P_{WT} + P_{PV} - L - P_{B,c} - P_{EL,c} \quad (19)$$

369 where,  $P_{B,c}$  and  $P_{EL,c}$  are the power required to fully charge the battery bank and hydrogen  
 370 tank respectively.  $P_{B,c}$  is given by:  
 371

$$372 \quad P_{B,c} = \frac{(1-SOC)n_B c_B V_B}{\eta_{B,c}} \quad (20)$$

373 where,  $\eta_{B,c}$  is the battery charging efficiency, and  $P_{EL,c}$  is given by:  
 374  
 375

376  
 377  
 378



$$P_{EL,c} = \min \left\{ P_{EL,nom}, \left( M_{H_2,max} - M_{H_2} \right) \frac{m_{H_2} LHV}{\eta_{EL}} \right\} \quad (21)$$

384

385 where,  $P_{EL,nom}$  and  $\eta_{EL}$  are the electrolyser nominal power and its efficiency, and  $M_{H_2,max}$   
 386 stands for the mass of the hydrogen in a fully charged hydrogen tank. The size of the  
 387 hydrogen tank,  $M_{H_2,max}$ , can be calculated using the nominal power of the fuel cell and an  
 388 autonomy period  $T_{a,H_2}$  (in days):

389

$$M_{H_2,max} = \frac{24 P_{FC,nom} T_{a,H_2}}{LHV \eta_{FC}} \quad (22)$$

391

392 The total annual sellable excess power is given by:

393

$$feed = \sum_{i=1}^{8760} P_{ex,i} \quad (23)$$

395

396 Or alternatively, the actual annual profit can be calculated if the feed-in tariff  $t_{grid}$  is given.  
 397 Since the feed-in tariff could have a variable rate depending on the time of the day (e.g.  
 398 cheaper rate at off-pick hours), the profit can be calculated as:

399

$$profit = \sum_{i=1}^{8760} [P_{ex} t_{grid}]_i \quad (24)$$

401

402

403 **Penetration**-System penetration,  $p$ , is defined as the ratio of the annual renewable power to  
 404 the annual demand load.

405

$$p = \frac{P_{R,t}}{L_t} = \frac{P_{WT,t} + P_{PV,t}}{L_t} \quad (25)$$

407

408 **CO<sub>2</sub> Emission**-The actual power produced by a diesel generator depends on its nominal  
 409 power, the demand load/power deficit and the operational scenarios set in the energy  
 410 management system. The hourly averaged diesel power  $P_D$  can get any value between 0 and  
 411  $P_{D,nom}$ . The total annual  $CO_2$  emission (in kg) is given by [46]:

412

$$CO_2 = \frac{0.246 \sum_{i=1}^{8760} P_{D,i} + 0.08145 P_{D,nom} T_D}{1000} e_l \quad (26)$$

414

415 where,  $T_D$  is the total number of hours of the operation per year,  $P_D$  is the hourly-averaged  
 416 diesel power (in W) and  $e_l$  is the emission of  $CO_2$  per litre of diesel consumption. This value  
 417 depends on both the diesel generator and the fuel characteristics and normally has a value  
 418 between 2.4 – 2.8 kg/l range [46]. In this study, we assume  $e_l = 2.68$  kg/l.

419

420 **Present Value of Total Life Span Cost (TLSC)** - The present value of the system cost over  
 421 its lifespan ( $TLSC$ ) is given by [20]:

422

$$TLSC = \sum_{j=0}^{Ns} \frac{C_j}{(1+d)^j} \quad (27)$$

424

425 in which,  $d$  stands for the annual discount rate,  $Ns$  is the lifespan of the system and  $C_j$  is the  
 426 cost in year  $j$ . The annual cost  $C_j$  includes the capital cost  $C_c$  for the year  $j = 0$  of operation,  
 427 the operation and maintenance costs  $C_{O\&M}$  and the replacement cost  $C_r$  for the years  $j =$

428 1, ...,  $N_S$  of operation. The capital cost of the system includes the initial cost  $C_i$  for buying the  
 429 system components and their installation cost  $C_{inst}$ . The installation cost for each component  
 430 is normally estimated as a fraction of the initial cost that component. The capital cost of the  
 431 whole system is, the summation of the initial cost  $C_i$  and installation cost  $C_{inst}$ , is therefore  
 432 given by:

$$433 C_c = \sum_{comp} C_{u,comp} S_{comp} (1 + \alpha_{inst,comp}) \quad (28)$$

436 in which,  $C_u$  and  $S$  are the unit cost and the size of component, and parameter  $\alpha_{inst}$  is the  
 437 fraction of the initial cost used for estimating the installation cost.

438  
 439 In cost analysis of HRES, only the cost of major components is considered and electronic and  
 440 energy management system is excluded. There are two reasons behind this: (i) the cost of  
 441 these components is much lower than the cost of other components and (ii) the cost of these  
 442 components is almost independent of the size of the system, hence, in HRES size  
 443 optimisation the inclusion or exclusion of these components does not affect the result of  
 444 optimisation.

445  
 446 The operation and maintenance cost of the system  $C_{O\&M}$  has two parts, namely, fixed  $C_{O\&M,F}$   
 447 and variable  $C_{O\&M,V}$  parts. Similar to installation cost the fixed part of O&M cost can be  
 448 estimated as a fraction of the initial cost using parameter  $\alpha_{O\&M}$ . In HRES, the only variable  
 449 part of O&M cost is the fuel cost, in case of having diesel generators in the system  
 450 configuration. The overall O&M cost of HRES is therefore given by:

$$451 C_{O\&M} = \sum_{comp} \alpha_{O\&M,comp} C_{i,comp} + C_{O\&M,V,D} \quad (29)$$

452 where,  $C_{O\&M,V,D}$  is the variable O&M cost of the diesel generator given by:

$$453 C_{O\&M,V,D} = \frac{0.246 \sum_{i=1}^{8760} P_{D_i} + 0.08145 P_{D,nom} T_D}{1000} C_{fuel} \quad (30)$$

454 in which,  $T_D$ , measured in hours, stands for total duration of operation of diesel generator  
 455 annually,  $P_{D_i}$  is the hourly-averaged diesel power, and  $C_{fuel}$  is the fuel price.

456  
 457 Components with a lifespan shorter than the desired lifespan of the system need to be  
 458 replaced throughout the duration of the system operation. The replacement cost of a  
 459 component  $C_{r,comp}$  depends on the number of replacements  $n_{r,comp}$  during the lifespan of the  
 460 system and the capital cost of that component  $C_{c,comp}$ . The overall replacement cost of a  
 461 HRES is therefore given by:

$$462 C_r = \sum_{comp} n_{r,comp} C_{c,comp} \quad (31)$$

463 The number of replacements of PV panels and wind turbines and are calculated based on  
 464 their nominal lifespan using the following equation:

$$465 n_{r,comp} = \left\lceil \frac{N_s}{N_{nom,comp}} \right\rceil \quad (32)$$

466  
 467  
 468

474 where,  $N_{nom,comp}$  is the nominal lifespan of PV panel  $N_{nom,PV}$  or wind turbine  $N_{nom,WT}$ .  
 475 Both of them, similar to the system lifespan  $N_S$ , are measured in years.

476  
 477 The number of replacements of fuel cell, electrolyser and diesel generator are calculated  
 478 based on the actual usage of these components and is given by:  
 479

$$480 \quad n_{r,comp} = \left\lceil \frac{N_S T_{comp}}{N_{comp,nom}} \right\rceil \quad (33)$$

481  
 482 where  $N_{comp,nom}$  and  $T_{comp}$  are, respectively, the nominal lifespan and the operating time,  
 483 both in hours.

484  
 485 The number of replacements for batteries,  $n_{r,B}$ , is calculated using the nominal lifespan of the  
 486 batteries and the equivalent life of the battery in years, whichever is shorter as given by [20]:  
 487

$$488 \quad n_{r,B} = \left\lceil \max \left\{ \frac{N_S}{N_{nom,B}}, \frac{N_S}{N_{eq,B}} \right\} \right\rceil \quad (34)$$

489  
 490 where,  $N_{eq,B}$  stands for the equivalent life of the battery and  $N_{nom,B}$  is the nominal life of the  
 491 battery, both measured in years. The nominal life of lead-acid batteries is about 4 years. The  
 492 equivalent life of the battery depends on the number and depth of charge-discharge cycles, as  
 493 given by:  
 494

$$495 \quad N_{eq,B} = \frac{1}{\sum_{k=1}^{n_d} \frac{1}{[n_{cycle\ to\ fail}]_k}} \quad (35)$$

496  
 497 where,  $n_d$  is the total number of charge-discharge cycles per year, and  $[n_{cycle\ to\ fail}]_k$ , the  
 498 number of cycles to failure, for lead-acid batteries is given by [20] as a function of depth of  
 499 discharge DOD:  
 500

$$501 \quad [n_{cycle\ to\ fail}]_k = 540.1 DOD_k^{-0.991} \quad (36)$$

502  
 503 Table 1 summarises the value of cost modelling parameters required for cost analysis of  
 504 HRES as reported in [46, 50, 51].  
 505  
 506

Table 1-Cost modelling parameters used in this study

	Wind turbine	PV panel	Battery	Diesel generator	Fuel cell	Electrolyser
$S$	Rotor area $A_{WT} (m^2)$	Panel area $A_{PV} (m^2)$	Battery bank capacity (Ah)	Nominal power $P_{D,nom} (W)$	Nominal power $P_{FC,nom} (W)$	Nominal power $P_{EL,nom} (W)$
$C_u$	Eq. 37	Eq. 38	Eq. 39	Eq. 40	4.08\$/ $W_{nom}$	2\$/ $W_{nom}$
$\alpha_{inst}$	0.2	0.4	0	0	0	0
$\alpha_{O\&M}$	0.03	0.01	0.01	0.15	0.1	0.1
$N_{nom}$	25 years	20 years	4 years	10,000 hours	5,000 hours	60,000 hours

507  
 508 For wind turbine:

$$509 \quad C_u (\$/m^2) = \begin{cases} 480 & A_{WT} > 1180 m^2 \\ -207 \times \log(A_{WT}) + 1944 & A_{WT} \leq 1180 m^2 \end{cases} \quad (37)$$

510

511 For PV panel:

512

$$513 \quad C_u (\$/m^2) = \begin{cases} 220 & A_{PV} > 1000 \text{ m}^2 \\ -51.64 \times \log(A_{PV}) + 580 & A_{PV} \leq 1000 \text{ m}^2 \end{cases} \quad (38)$$

514

515 For battery bank, the unit cost depends on the battery capacity,  $c_B$ , and a discount rate for  
516 bulk purchase:

517

$$518 \quad C_u (\$/Ah) = 163c_B^{-1.14} \times \begin{cases} 0.8 & n_B > 100 \\ -0.0015n_B + 0.95 & n_B \leq 100 \end{cases} \quad (39)$$

519

520 For diesel generator:

521

$$522 \quad C_u (\$/W) = \begin{cases} 0.4 & P_{D,nom} > 50000 \text{ W} \\ 1.7 \times 10^{-10}P_{D,nom}^2 - 1.84 \times 10^{-5}P_{D,nom} + 0.8971 & P_{D,nom} \leq 50000 \text{ W} \end{cases} \quad (40)$$

523

524

525 **Levelised Cost of Energy**-Levelised cost of energy is calculated based on the usable power  
526 produced by the system and the cost of production within a year. The annualised cost of the  
527 system,  $C_a$ , is determined by:

528

$$529 \quad C_a = TLSC \frac{d(1+d)^{Ns}}{(1+d)^{Ns}-1} \quad (41)$$

530

531 where, the fraction in this equation is the uniform capital recovery factor. For standalone  
532 HRES, the LCE of a system,  $C_l$  is given by:

533

$$534 \quad C_l = \frac{C_a}{P_t} \quad (42)$$

535

536 where,  $P_t$  is the annual usable power output of the system. It should be noted that for  
537 standalone HRES the output power is not necessarily entirely usable. The produced power  
538 excess to the load is used for charging the battery bank and hydrogen tank. However, if the  
539 excess power is more than the required power for charging the battery bank and hydrogen  
540 tank, the difference will be dumped. Hence, the usable annual output for standalone HRES is:

541

$$542 \quad P_t = L_t - U_t \quad (43)$$

543

544 in which,  $L_t$  and  $U_t$  are the total annual demand load and the total annual unmet load  
545 respectively.

546

547 For on-grid HRES, the unmet load is compensated by the grid. Therefore, the LCE has two  
548 terms, namely the cost of producing energy by the HRES and the cost of buying electricity  
549 from the grid to compensate for the unmet load:

550

$$551 \quad C_l = \frac{C_a}{P_t} + \frac{C_{u,t}}{P_{u,t}} \quad (44)$$

552

553 The cost of electricity from grid depends on the unmet load and, in the case of variable rates,  
554 the time of the day. The total cost of electricity from the grid is therefore given by:

555  
556  $C_{u,t} = \sum_{i=1}^{8760} [P_u C_{grid}]_i$  (45)

557  
558 **2.4 Optimisation Formulation**

559 In view of the design variables and design qualities defined in the previous two subsections,  
560 the optimisation problem can be formulated as:

561  
562  $\min \vec{Y}_1(\vec{X})$  and  $\max \vec{Y}_2(\vec{X})$  (46.a)

563 *s. t.*

564  $\vec{Y}_3 \leq \vec{Y}_{3,c}$  (46.b)

565  $\vec{Y}_4 \geq \vec{Y}_{4,c}$  (46.c)

566  $\vec{X}^l \leq \vec{X} \leq \vec{X}^u$  (46.d)

567  
568 where,  $\vec{X} = \{n_{WT}, A_{WT}, A_{PV}, n_B, P_{D,nom}, P_{FC,nom}, P_{EL,nom}\}$  is the vector of design variables,  
569 with  $\vec{X}^l$  and  $\vec{X}^u$  as given by Equations 2 through 9 for the generic configuration, or as  
570 explained above for special cases of retrofitting, fixed configuration and pre-sized  
571 components. Vectors  $\vec{Y}_1$  and  $\vec{Y}_2$ , respectively, contain the objectives to be minimised and  
572 maximised. Vectors  $\vec{Y}_3$  and  $\vec{Y}_4$  contains those design qualities which are subjected to the  
573 constraints  $\vec{Y}_{3,c}$  and  $\vec{Y}_{4,c}$ .

574  
575 These four vectors are disjoint subsets of design qualities:

576  
577  $\vec{Y}_1 \cup \vec{Y}_2 \cup \vec{Y}_3 \cup \vec{Y}_4 \subseteq \{TLSC, LCE, U_t, CO_2, p, feed, profit\}$  (47.a)

578  
579  $\vec{Y}_1 \cup \vec{Y}_3 \subseteq \{TLSC, LCE, U_t, CO_2\}$  (47.b)

580  
581  $\vec{Y}_2 \cup \vec{Y}_4 \subseteq \{p, feed, profit\}$  (47.c)

582  
583 **3 Optimisation Method**

584 The optimisation methods must be able to search the design space at two levels, namely,  
585 configuration level and size level. In this study, we use a GA and a nondominated sorting GA  
586 (NSGA-II) for single and multiobjective optimisation respectively. The integrated  
587 configuration-size formulation and the single and multiobjective optimisation algorithms  
588 have been implemented in the software tool MOHRES (Multiobjective Optimisation of  
589 Hybrid Renewable Energy Systems) [52, 53]. Figure 1 shows the dataflow in MOHRES.  
590

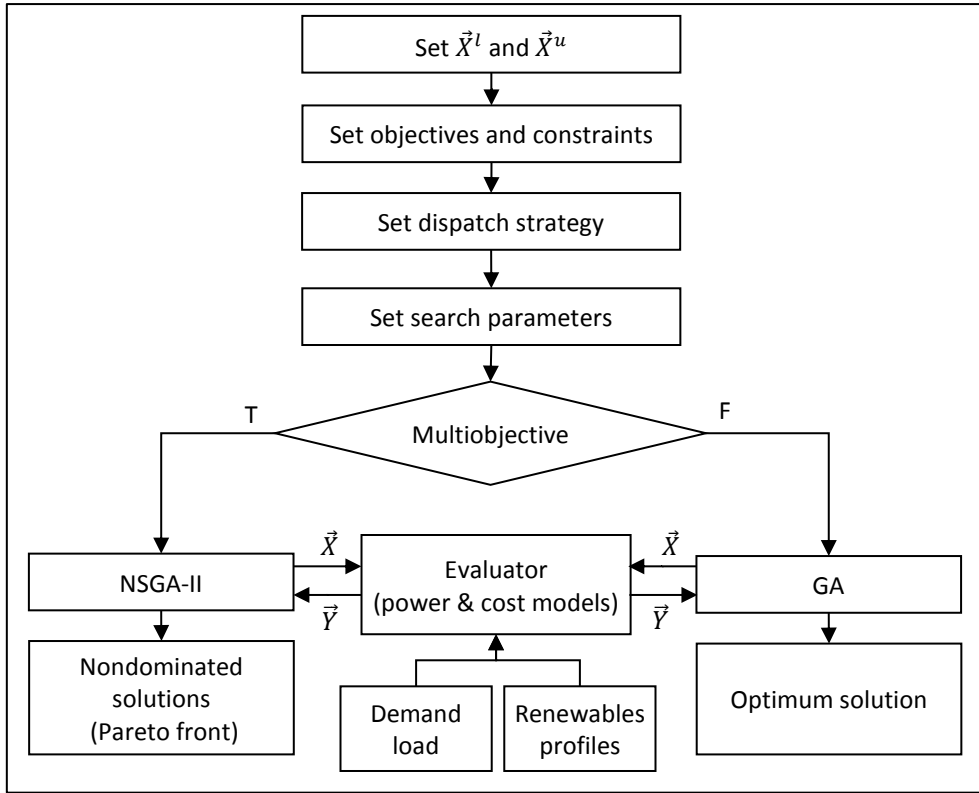


Figure 1-Iterated configuration-size optimisation dataflow in MOHRES

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The upper and lower limits of the design variables are set using Equations 2-9 for a full integrated configuration-size optimisation, and a combination of these equations and fixed values for special cases of optimisation for retrofitting and fixed configuration. Objectives and constraints are set according to the optimisation problem 46 and 47.

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In configurations which include more than one storage/auxiliary component different dispatch strategies can be defined. A dispatch strategy is defined based on the charging and usage orders of the storage/auxiliary components. For instance, in a wind-PV-fuel cell-electrolyser-diesel-battery bank configuration, 6 different usage orders can be defined based on the precedence of the battery bank, diesel generator and fuel cell in compensating the power deficit. For the same configuration, 2 charging orders can be defined based on the precedence of the battery bank and hydrogen tank in charging where there is an excess power [26].

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Generally speaking, a search mechanism should be able to deliver both exploration and exploitation at all levels. However, here at the configuration level we are dealing with a highly discrete and small design space (of around twenty-odd members for the problem at hand). Therefore, exploitation does not apply at configuration level. The reproduction operators, crossover and mutation, therefore must be designed in such a way that collectively provide exploration and exploitation at size level and exploration at configuration level.

615  
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617

Arithmetic weighted crossover is a natural choice for crossover mechanism in sizing problems. A weighted arithmetic crossover is therefore used for exploitation of the design space at size level. Each crossover operation generates two offspring as defined by:

618

$$\vec{X}_{child,1} = \lambda \vec{X}_{parent,1} + (1 - \lambda) \vec{X}_{parent,2} \quad (48.a)$$

619

620

$$621 \quad \vec{X}_{child,2} = \lambda \vec{X}_{parent,2} + (1 - \lambda) \vec{X}_{parent,1} \quad (48.b)$$

622

623 where,  $\lambda \in (0,1)$  is a random number.

624

625 The arithmetic crossover mechanism defined above, has some exploration capability at the  
626 configuration level: by combining two parents of different configurations (combining of two  
627 chromosomes with some zero genes placed at different locations), there is a chance that the  
628 offspring have a configuration with more components (a chromosome with more non-zero  
629 genes) than those of its parents. For example, a crossover between a wind-PV solution and a  
630 PV-battery-diesel solution will lead to at least one wind-PV-battery-diesel solution  
631 (depending on the value of the random number  $\lambda$  in Equations 48). The exploration at  
632 configuration level due to arithmetic crossover is a one-way ‘upward’ exploration. That is, it  
633 can only produce offspring with similar or more populated configurations than those of  
634 parents. An upward/downward change in configuration refers to adding/removing  
635 components to/from the system configuration.

636

637 The mutation operator is therefore designed in such a way that besides exploration at size  
638 level, conduct a ‘downward’ exploration at configuration level too. At configuration level, a  
639 number of randomly selected genes in a randomly selected solution are set to zero. This is  
640 equivalent to removing the corresponding components from the configuration (downward  
641 exploration). Mutation at configuration level operates as:

642

$$643 \quad \vec{X}_{mute} = \vec{M}_1 \circ \vec{X}_{parent} \quad (49)$$

644

645 where,  $\vec{M}_1$  is a  $1 \times n_x$  mask vector with randomly generated 0 and 1 entries, the operator ‘ $\circ$ ’  
646 is Hadamard (entry-wise) product and  $n_x$  is the number of design variables in the  
647 chromosome (here 7). As a result of the mutation operation above, the design variables  
648 corresponding to the zero elements of the mask vector  $\vec{M}_1$  are set to zero. This is equivalent to  
649 the exclusion of the corresponding components from the configuration and therefore a  
650 downward exploration in the configuration.

651

652 At size level, a dynamic mutation operator is designed with two functions of exploration at  
653 earlier generations and exploitation at later generations. Although it is very unusual to use  
654 mutation operator for exploitation, it is shown later that adding exploitative functionality to  
655 the mutation operator will boost the performance of the search significantly. A selected  
656 solution is perturbed according to:

657

$$658 \quad \vec{X}_{mute} = \vec{X}_{parent} + \delta \vec{X} \quad (50)$$

659

660 where, the perturbation vector  $\delta \vec{X}$  is a randomly generated vector in the neighbourhood of the  
661 parent. The radius of the neighbourhood shrinks with a factor of  $\left(1 - \frac{i_{gen}-1}{n_{gen}-1}\right)$ , in which,  $n_{gen}$   
662 is the total number of generations and  $i_{gen}$  ( $1 \leq i_{gen} \leq n_{gen}$ ) is the current generation  
663 number. This allows fine-tuning (exploitation) at higher generations.

664

665 The  $i$  –  $th$  element of the perturbation vector  $\delta \vec{X}$ ,  $\delta X_i$ , is a random value selected from the  
666  $i$  –  $th$  row of the neighbourhood matrix  $N_{mute}$  defined as:

667  
668  $N_{mute} = M_2 \circ I_{mute}$  (51)  
669

670 in which,  $M_2$  is an  $n_x \times 2$  mask matrix with randomly generated 0 and 1 entries and  $I_{mute}$  is  
671 the shrinking mutation neighbourhood:  
672

673  $[I_{mute}]_{n_x \times 2} = \left(1 - \frac{i_{gen-1}}{N_{gen-1}}\right) \begin{bmatrix} \vec{X}^l - \vec{X}_{parent} \\ \vec{X}^u - \vec{X}_{parent} \end{bmatrix}^T$  (52)  
674

675 The mask matrix  $M_2$  is responsible for selecting the genes going through mutation at size  
676 level and the direction of the perturbation. For instance, a randomly generated matrix  $M_2 =$   
677  $\begin{bmatrix} 0 & 0 & 1 & 1 & 0 & 1 & 0 \\ 0 & 1 & 1 & 0 & 0 & 0 & 0 \end{bmatrix}^T$  excludes  $X_1$ ,  $X_5$  and  $X_7$  (the number of wind turbines, and the  
678 size of diesel generator and electrolyser) from mutation (no perturbation for these design  
679 variables), allows a positive perturbation for  $X_2$  (the size of wind turbine), a positive or  
680 negative perturbation for  $X_3$  (the size of PV panels), and a negative perturbation for  $X_4$  and  
681  $X_6$  (the size of battery bank and fuel cell).  
682

683 While the parent selection for the mutation at configuration level takes place randomly, the  
684 parent selection at the size level is based on a combination of a random selection and a  
685 fitness-based selection. The classical random selection is used where the population is  
686 diverse. The parent selection will be switched to a fitness based, where a roulette wheel is  
687 used to give higher chance of fine-tuning to the fitter solutions, as the diversity of the  
688 population decreases towards the end of the search (where the average fitness  $fit_{av}$   
689 approaches the maximum fitness  $fit_{max}$ ). In this study the limit of  $\frac{fit_{av}}{fit_{max}} \geq 0.9$  is used as  
690 the limit for fine-tuning (switching the parent selection from random to roulette wheel based).  
691

692 Another uncommon feature of the mutation operator designed for the problem at hand is  
693 related to the probability of mutation  $P_m$ . A dynamic  $P_m$  is defined as follows:  
694

695  $P_m = P_{m,config} + P_{m,size}$  (53)  
696

697 where,  $P_m$ ,  $P_{m,config}$  and  $P_{m,size}$ , respectively, represent the total mutation rate, the fraction  
698 of the mutations applied at the configuration level (Equation 49), and the fraction of the  
699 mutations applied at the size level (Equation 50):  
700

701  $P_{m,config} = 0.5P_{m,0} \left(1 - \frac{i_{gen-1}}{n_{gen-1}}\right)$  (54)  
702

703  $P_{m,size} = 0.5P_{m,0}$  (55)  
704

705 The total mutation rate  $P_m$  ( $0.5P_{m,0} \leq P_m \leq P_{m,0}$ ) decreases as the generation number  $i_{gen}$   
706 increases. The limits  $P_{m,0}$  and  $0.5P_{m,0}$  are associated to  $i_{gen} = 1$  and  $i_{gen} = n_{gen}$   
707 respectively. A relatively high value for  $P_{m,0} \sim 0.5 - 0.9$  should be used in order to allow  
708 effective mutation operation at both size and configuration levels.  
709

710 It should be noted that, although the design variables are treated as continuous real numbers,  
711 for practicality reasons, the design variables are rounded.  $A_{PV}$  and  $n_B$  are rounded up to the



712 nearest integer number, the nominal powers (diesel, fuel cell and electrolyser) are rounded up  
 713 to the nearest 100 watt, and the wind turbine rotor radius  $R_{WT}$  is rounded up to the nearest 10  
 714 cm.

715  
 716 In the problem formulation of (46) and (47) above, there is no limitation on the number of  
 717 constraints. Hence, the initial population generation for highly constrained optimisation  
 718 problems may take a very long time if infeasible solutions are rejected. In such cases, the  
 719 crossover and mutation may also become ineffective due to generating solutions with high  
 720 chance of being rejected due to infeasibility. Hence, penalising infeasible solutions is the best  
 721 approach in these cases. On the other hand, for flexible design spaces, adopting a rejection  
 722 method leads to starting from a feasible set of solutions which is more likely to get to the  
 723 global optima in a smaller number of generations. Therefore, two methods of rejection and  
 724 penalising infeasible solutions are used for different cases, depending on the number of  
 725 constraints and the rigidity of the design space.

726  
 727 Equation 56 shows a generalised fitness function with a great flexibility, which is developed  
 728 for single objective optimisation using GA. This form can be used for both maximisation and  
 729 minimisation cases and can take the form of a raw fitness as well as the form of a penalised  
 730 fitness, by setting some coefficients by the user.

$$731 \quad 732 \quad fit = (fit_r \prod_{i=1}^{n_c} P_i)(1 - fit_0) + fit_0 \quad (56)$$

733  
 734 In this equation,  $fit$  stands for the fitness,  $fit_r$  is the normalised raw fitness,  $n_c$  is the number  
 735 of constraints,  $fit_0$  is the worst possible normalised fitness (set by the designer), and  
 736  $0 < P_i \leq 1$  is the penalty value for the  $i$ -th constraints. A value of  $fit_0 = 0.1$  is used for the  
 737 single objective optimisation cases studies in this paper. By doing this, we prevent assigning  
 738 very small fitness values to bad solutions, and therefore, any potentially good information  
 739 from the worst solution can still be retrieved in crossover operation.

740  
 741 The raw fitness for a solution is calculated as follows:  
 742

$$743 \quad fit_r = \begin{cases} \frac{f_n}{f+f_n} & \text{when } f \text{ is to be minimised} \\ \frac{f}{2f_n} & \text{when } f \text{ is to be maximised} \end{cases} \quad (57)$$

744  
 745 where  $f$  is the calculated objective function for that solution, and  $f_n$ , the average value of the  
 746 objective function of solutions in the initial population, is used to normalise the fitness.

747  
 748 Here, the infeasibility is measured in terms of both the number of contradicted constraints  
 749 and the amount of deviation from the constraint limits. The penalty applied to each  
 750 contradicted constraint is calculated by:

$$751 \quad 752 \quad P_i = \exp(-\mu_i \delta_i) \quad (58)$$

753  
 754 where,  $\delta_i$  is the deviation from the constraint limit and  $0 \leq \mu_i$  is the strength of the applied  
 755 penalty on the constraint  $i$ . Parameter  $\mu_i = 0$  represents the case of no penalty if the  
 756 constraint  $i$  is contradicted. Deviations  $\delta_i$  are normalised values and given by:  
 757

$$\delta_i = \begin{cases} 0 & \text{satisfied inequality constraint} \\ \left| \frac{Y_i - Y_{c,i}}{\max\{Y_{c,i}, Y_i\}} \right| & \text{contradicted inequality constraint} \end{cases} \quad (59)$$

759

760 where,  $Y_i$ 's are the entries of  $\vec{Y}_3 \cup \vec{Y}_4$  and  $Y_{i,c}$ 's are the entries of  $\vec{Y}_{3,c} \cup \vec{Y}_{4,c}$  as defined in the  
 761 optimisation problem of (46) and (47). The normalisation term  $\max\{Y_{c,i}, Y_i\}$  is always  
 762 nonzero for any contradicted constraint, irrespective of the type of inequality constraint and  
 763 the values  $Y_{c,i}$  and  $Y_i$ .

764

765 The goal of NSGA-II (nondominated sorting genetic algorithm) is to find the best estimation  
 766 of the Pareto front solutions. This is done by generating a set of nondominated solutions and  
 767 moving it forward towards the actual Pareto front generation by generation. A successful  
 768 NSGA-II should be aiming at producing enough and uniformly distributed solutions on the  
 769 Pareto front. In order to improve the first front generation by generation, the same  
 770 reproduction operators (crossover and mutation) as those defined for single objective GA can  
 771 be used. In single objective GA, the parent selection for crossover is solely based on a  
 772 roulette wheel, which is formed based on the fitness of the solutions in the population. In  
 773 multiobjective NSGA-II, the parent selection for crossover is different. It is based on a  
 774 tournament selection mechanism which is based on (i) the rank of a solution (the front that  
 775 the solution belongs to) and (ii) the crowding distance. The crowding distance is a measure of  
 776 how close a solution is to its neighbours on the same front. Both are important qualities. The  
 777 rank of a solution is important as solutions on the low-rank fronts are more likely to produce  
 778 better offspring, if selected as a parent. The crowding distance is also important as those  
 779 solutions located in less crowded regions, if selected as a parent for crossover, are more likely  
 780 to produce offspring in their own neighbourhood and therefore filling the gaps on the front.

781

782 The common search parameters for both GA and NSGA-II are:  $P_c$ ,  $P_{m,0}$ ,  $n_{pop}$  and  $n_{gen}$ . For  
 783 GA, the method of handling of infeasible solutions (rejection or penalty) is also a search  
 784 parameter, and so are the penalty weights  $\mu_i$  in case of using penalty method. For any  
 785 generated design candidate within the GA or NSGA-II search (produced either randomly as  
 786 in initial population generation or as a result of the mutation and crossover using Equations  
 787 48-55), the evaluator takes the system size  $\vec{X}$ , the demand load and renewables profiles and  
 788 uses the power and cost models (Equations 10-45) to find the relevant qualities of the design  
 789 candidate  $\vec{Y}$  (objectives and constrained design qualities as defined in 47). These  $\vec{Y}$  values  
 790 are returned to the search algorithm for assigning fitness and rank for that individual  
 791 according to Equations 56-59.

792

#### 793 **4 Case Studies**

794 In this section, 8 case studies, denoted by CS1 to CS8, are reported. These case studies are  
 795 designed carefully to assess the robustness of the presented problem formulation in solving  
 796 different types of HRES optimisation problems and to evaluate the performance of the  
 797 developed GA and NSGA-II for integrated configuration-size optimisation.

798

799 Case studies are summarised in Table 2. The first six case studies are single objective and the  
 800 last two are multiobjective. Three sites are used for these case studies. The hourly averaged  
 801 demand load and the renewable resources for four seasonal typical days of Site 1 are given in  
 802 the appendix. Site 2 has similar demand load profile as that of Site 1 but the solar irradiance  
 803 and wind speed are, respectively, 20% lower and 50% higher than of those in Site 1. Site 3 is  
 804 a low-renewable site. The demand load and wind profiles are identical to Site 1, but the solar

805 irradiance is half of that of Site 1. Both standalone (S) and grid connected cases (G) are  
 806 considered.

807

808 The purpose of CS1 is to show how the presented problem formulation starts with a generic  
 809 all-components-in configuration and ends with an optimum configuration by eliminating cost  
 810 intensive components. The optimisation problem of CS2 is identical to that of CS1 but the  
 811 renewable resource profile is different. This case study is planned to show how the integrated  
 812 configuration-size optimisation problem formulation finds a different optimum configuration  
 813 as the renewable resources' profiles change. CS3 is defined to show how adding a new  
 814 constraint influences the optimum configuration. In CS4 the optimum size of components for  
 815 a predefined configuration are obtained. This case study is planned to show the flexibility of  
 816 the problem formulation in dealing with fixed configurations (reducing from integrated  
 817 configuration-size optimisation to size optimisation only). CS5 and CS6 are aimed at  
 818 showing the flexibility of the formulation in delivering special cases of retrofitting of an  
 819 existing power system by adding new components to an existing system. In CS5, the existing  
 820 power system is a 5 kW (*nominal*) diesel generator. This diesel generator produces 37  
 821 tonnes of CO<sub>2</sub> emission per year when operating continuously. The purpose of the  
 822 optimisation is to find the best retrofitted configuration that includes this diesel generator (for  
 823 saving) but reduces its CO<sub>2</sub> emission to less than 500 kg per year (for environment). CS6  
 824 presents another retrofitting case but starts with a different existing power system (a  
 825 30kW @  $V_{rated} = 9\text{ m/s}$  wind turbine with a rotor radius of 6.9 m) and aims at a fixed  
 826 retrofitted configuration (wind-PV-battery). The intention for this retrofitting is to invest in  
 827 renewable energy by selling the excess power to the grid.

828  
 829

Table 2-Summary of Case Studies

Case study	Configuration	Obj.	Const.	Site	Standalone/ on grid
CS1	Generic; $\vec{X}^l$ Eq. 2; $\vec{X}^u$ Eq.s 3 to 9	<i>LCE</i>	$U_t = 0$	1	S
CS2	Generic; $\vec{X}^l$ Eq. 2; $\vec{X}^u$ Eq.s 3 to 9	<i>LCE</i>	$U_t = 0$	2	S
CS3	Generic; $\vec{X}^l$ Eq. 2; $\vec{X}^u$ Eq.s 3 to 9	<i>LCE</i>	$U_t = 0$ $p > 200\%$	2	S
CS4	Fixed: W-FC-EL $\vec{X}^l$ Eq. 2; $\vec{X}^u =$ $\{R_{WT}^u, 0, 0, 0, P_{FC,nom}^u, P_{EL,nom}^u\}$ & Eq.s 4, 7 and 8	<i>LCE</i>	$U_t = 0$	1	S
CS5	Retrofitting: diesel only to generic config. $\vec{X}^l = \{0, 0, 0, P_{D,nom}, 0, 0\}$ $\vec{X}^u = \{R_{WT}^u, A_{PV}^u, n_B^u, P_{D,nom}, P_{FC,nom}^u, P_{EL,nom}^u\}$ & Eq.s 4, 5 and 7 through 9 $P_{D,nom} = 5kW$	<i>TLSC</i>	$U_t = 0$ $CO_2 < 500\text{ kg}$	1	S
CS6	Retrofitting: wind only to fixed config. W-PV-B $\vec{X}^l = \{R_{WT}, 0, 0, 0, 0, 0\}$ $\vec{X}^u = \{R_{WT}, A_{PV}^u, n_B^u, 0, 0, 0\}$ & Eq.s 5 and 9 $R_{WT} = 6.9\text{ m}$	<i>profit</i>	$U_t = 0$ $TLSC \leq 250,000$	1	G
CS7	Generic; $\vec{X}^l$ Eq. 2; $\vec{X}^u$ Eq.s 3 to 9	<i>LCE</i> $U_t$	$p \geq 60\%$	1	S
CS8	Generic; $\vec{X}^l$ Eq. 2; $\vec{X}^u$ Eq.s 3 to 9	<i>LCE</i> $U_t$ $CO_2$		3	S

830

831 In single objective optimisation problems, we are looking for a solution with best  
832 performance in terms of one of the design qualities and have ‘hard’ constraints on some other  
833 design qualities. For instance, in CS1 we are looking for the cheapest fully reliable system  
834 (min  $LCE$  subject to  $U_t = 0$ ). From a practical point of view, this is not always the case. We  
835 can often tolerate some inferior performance measures as we know that losing on one quality  
836 leads to a gain on another quality (conflicting objectives). In cases like this conducting a  
837 multiobjective optimisation is the best way to get an insight to the interaction between  
838 different objectives. Once a Pareto front is obtained the designer selects a solution through a  
839 trade-off study. Case studies CS7 and CS8 are multiobjective problems. In CS7 we deal with  
840 two conflicting cost-related and reliability-related objectives:  $LCE$  and  $U_t$ . CS8 is defined for  
841 a low-renewables site with three conflicting objectives:  $LCE$ ,  $U_t$  and the third one related to  
842 the impact of the system on the environment:  $CO_2$  emission.

843

844 In all case studies the usage order is defined as: battery first, then fuel cell (hydrogen), and  
845 then diesel. The charging order is defined as: battery first then hydrogen tank. All power and  
846 cost model parameters which have not been defined within the text are given in Table A1 in  
847 the appendix.

848

849 For consistency the same set of search parameters:  $P_c = 0.3$ ,  $P_{m,0} = 0.9$  and  $n_{gen} = 100$   
850 are used for both single objective and multiobjective cases. Population size  $n_{pop}$  is set to 20  
851 and 40 for GA and NSGA-II respectively.

852

#### 853 4.1 Results and Discussion

854

855 The optimum solutions for the first six case studies are summarised in Table 3, followed by  
856 detailed discussion including the performance of the GA for each case study.

857

858 Table 3-Optimum solutions for single objective case studies CS1 through CS6

Case study	$\vec{X}_{opt}$							$\vec{Y}_{opt}$					
	$n_{WT}$	$R_{WT}$ ( $m^2$ )	$A_{PV}$ ( $m^2$ )	$n_B$	$P_{D,nom}$ ( $W$ )	$P_{FC,nom}$ ( $W$ )	$P_{EL,nom}$ ( $W$ )	$TLSC$ (1000\$)	$LCE$ ( $c/kWh$ )	$U_t$ ( $kW$ )	$p$ (%)	$CO_2$ ( $kg$ )	$profit/feed$ (1000\$/MW)
CS1	1	0	296	232	0	0	0	186	23.0	0	139	0	N/A
CS2	1	4.7	0	14	8300	0	0	186	23.1	0	134	5700	N/A
CS3	1	5.7	14	62	0	0	0	192	23.9	0	201	0	N/A
CS4	1	11.6	0	0	0	8500	4800	694	86.2	0	378	0	N/A
CS5	1	0	285	232	5000	0	0	192*	23.8*	0	134	400	N/A
CS6	1	6.9	372	112	0	0	0	250**	31.0**	0	284	0	(12.8/106.7)

\*Excluding the capital cost of the diesel generator  
\*\*Excluding the capital cost of the wind turbine

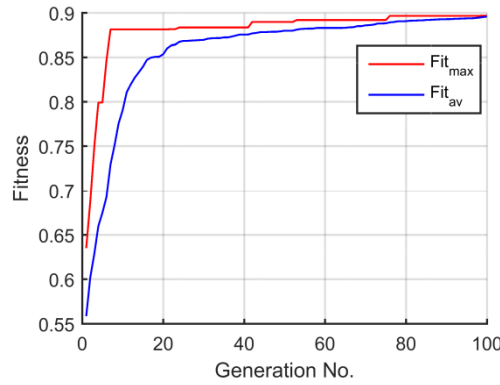
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#### 860 CS1: Configuration-size optimisation-Site 1

861 A closer look at the renewable resources of the appendix suggests that this site has low-wind  
862 and high-solar irradiance profiles. In view of this, although starting from a generic wind-PV-  
863 battery-diesel-Fuel cell/electrolyser, a PV-battery configuration is the optimum configuration  
864 that we expect to see as a result of an integrated configuration-size optimisation for CS1.

865 Starting from a generic configuration and ending with a PV-battery configuration implies that  
866 wind turbine, diesel generator and fuel cell/electrolyser have been removed from the  
867 configuration at some points during the search process. Figure 2 shows the GA search history  
868 for this case study and Table 4 shows the configuration and components’ size of the best  
869 solution in each generation. The last column of Table 4 shows the change in the

870 configuration, where, as mentioned before downward and upward directions stand for  
 871 removing and adding components to the system respectively.



872  
 873 Figure 2-GA search history for CS1  
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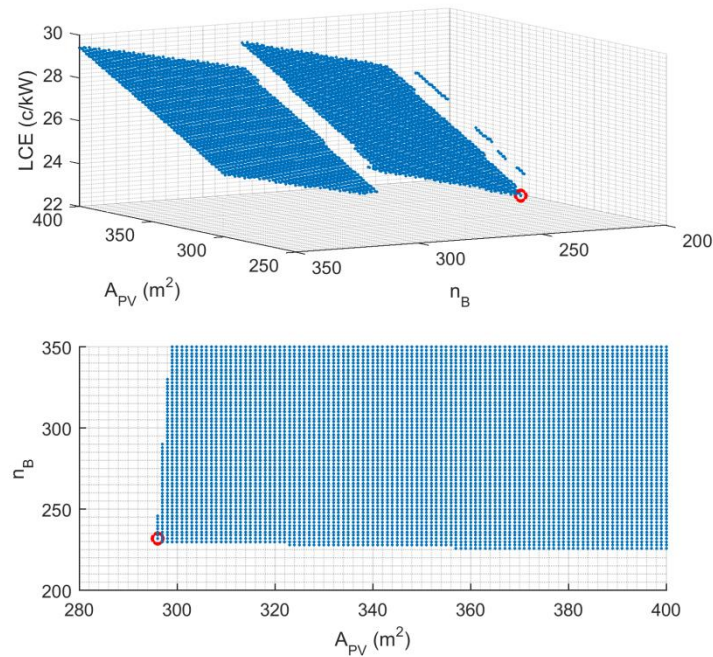
875 The final configuration is due to a combination of a number of downward and upward  
 876 changes as shown in Table 4. The best solution of the second generation ( $i_{gen} = 2$ ), is the  
 877 result of a downward mutation of one of the solutions in the first generation, which has led to  
 878 the removal of the fuel cell (or electrolyser) and consequently the accompanied electrolyser  
 879 (fuel cell). A further downward mutation leads to the removal of the wind turbine and diesel  
 880 at  $i_{gen} = 6$ . An upward mutation or crossover brings back the diesel generator into the  
 881 configuration at  $i_{gen} = 76$ , which is removed again later as a result of another downward  
 882 mutation at  $i_{gen} = 99$ . Having both downward and upward changes in the configuration  
 883 proves the capability of the developed algorithm in exploring various configurations while  
 884 optimising the size of components. Exploration at configuration level at both directions  
 885 shows that the algorithm does not get trapped in a local optima.  
 886  
 887

888 Table 4- System configuration in the search history of CS1

$i_{gen}$	$fit_{max}$	$fit_{av}$	$R_{WT}$ ( $m^2$ )	$A_{PV}$ ( $m^2$ )	$n_B$	$P_{D,nom}$ (W)	$P_{FC,nom}$ (W)	$P_{EL,nom}$ (W)	Direction of change in configuration
1	0.6352	0.5586	2	339	278	43400	47900	52000	no change
2	0.6876	0.6021	10.9	122	772	32300	0	0	downward
3	0.7494	0.6288	8	219	734	12800	0	0	no change
4	0.7992	0.6602	4.5	422	552	6100	0	0	no change
5	0.7992	0.6751	4.5	422	552	6100	0	0	no change
6	0.8465	0.6929	0	314	832	0	0	0	downward
7	0.8815	0.7293	0	356	282	0	0	0	no change
...									no change
75	0.8920	0.8883	0	323	236	0	0	0	no change
76	0.8967	0.8894	0	297	234	100	0	0	upward
...									no change
98	0.8967	0.8947	0	297	232	100	0	0	no change
99	0.8968	0.8953	0	296	232	0	0	0	downward
100	0.8968	0.8960	0	296	232	0	0	0	no change

888  
 889 With reference to Table 4, one can observe that in earlier generations, changes in  
 890 configuration leads to jumps in  $fit_{max}$ . However, in later generations changes in  
 891 configuration are more like fine-tuning of configuration (i.e. adding a very small diesel of  
 892 100 W to the system at  $i_{gen} = 76$  and then removing it at  $i_{gen} = 99$ ) and have smaller effect  
 893 on maximum fitness.  
 894

895 For this particular case study, in which the optimum solution includes only two components  
 896 in the configuration, we can prove the optimality of the obtained sizes by employing a refined  
 897 exhaustive search in the neighbourhood of the solution. Using a grid size of 1 for both  $A_{PV}$   
 898 and  $n_B$ , within the limits:  $200 \leq A_{PV} \leq 400$  and  $150 \leq n_B \leq 350$ , over 40000 system  
 899 analyses are required. The results are shown in Figure 3. The solution identified by red circle  
 900 has the minimum LCE while satisfying the constraint of  $U_t = 0$ . This is identical to the  
 901 optimum solution obtained by the GA ( $A_{PV} = 396 \text{ m}^2$ ;  $n_B = 232$ ).  
 902



903 Figure 3-Exhaustive search results; top: feasible solutions; bottom: feasible design space  
 904  
 905

906 It should be noted that, although for this case study the optimality of the PV-battery  
 907 configuration could be argued with confidence with reference to the renewable resource  
 908 profiles, there is no practical means to prove it. If we were to use an exhaustive search, even  
 909 by assuming a very coarse grid of only 100 points for every one of the six design variables,  
 910  $10^{12}$  system evaluations over the entire of the domain would be required.  
 911

### 912 CS2: Configuration-size optimisation -Site 2

913 In comparison to the renewable resources of CS1, in CS2 the wind is 50% stronger but the  
 914 solar irradiance is 20% weaker. This makes, wind energy more cost effective and solar  
 915 energy more cost intensive. Therefore, compared to the optimum configuration of CS1, we  
 916 expect to see an increase in the share of power production by wind and a reduction of the  
 917 contribution of PV. In fact, in this case study, the PV has been eliminated from the generic  
 918 configuration. The presence of diesel in the configuration can be explained in view of the fact  
 919 that, unless the site has high renewable resource profiles, with the current prices (the cost  
 920 parameters used in this study) the LCE produced by diesel generator is still less than the LCE  
 921 produced by renewables. Figure 4 shows the GA search history for this case study and Table  
 922 5 shows the configuration history. This table shows a number of bidirectional changes in the  
 923 configuration ( $i_{gen} = 2, 4, 8$  and  $16$ ) as well as downward and upward changes ( $i_{gen} = 21,$   
 924  $35$  and  $36$ ). For instance the configuration of  $i_{gen} = 21$  has simultaneously gone through a  
 925 downward mutation, which has led to the removal of the wind turbine, and an upward

926 mutation or crossover which has brought back the diesel into the configuration. A closer look  
 927 at the configurations obtained by bidirectional changes, one can see that bidirectional changes  
 928 correspond to the removal/adding of a renewable component and adding/removal of a storage  
 929 component.

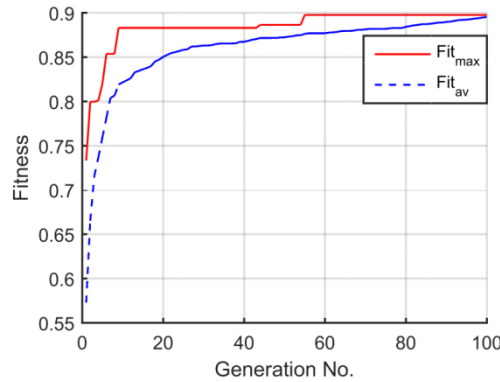


Figure 4-GA search history for CS2

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Table 5- System configuration in the search history of CS2

$i_{gen}$	$fit_{max}$	$fit_{av}$	$R_{WT}$ ( $m^2$ )	$A_{PV}$ ( $m^2$ )	$n_B$	$P_{D,nom}$ ( $W$ )	$P_{FC,nom}$ ( $W$ )	$P_{EL,nom}$ ( $W$ )	Direction of change in configuration
1	0.681	0.557	3.1	460	446	93600	15900	14600	no change
2	0.756	0.597	5.6	349	56	102800	0	0	downward
3	0.756	0.633	5.6	349	56	102800	0	0	no change
4	0.790	0.681	0	459	260	102800	0	0	downward
...									no change
7	0.813	0.730	0	402	236	80400	0	0	no change
8	0.832	0.749	4.2	277	80	13000	0	0	upward
...									no change
15	0.832	0.802	4.2	277	80	13000	0	0	no change
16	0.835	0.807	6	21	418	0	0	0	downward
...									no change
20	0.835	0.814	6	21	418	0	0	0	no change
21	0.845	0.821	0	385	220	1300	0	0	downward and upward
...									no change
34	0.845	0.834	0	385	220	1300	0	0	no change
35	0.850	0.836	2.3	349	288	0	0	0	downward and upward
...									no change
38	0.850	0.838	2.3	349	288	0	0	0	no change
39	0.854	0.839	5.3	0	150	5800	0	0	downward and upward
...									no change
100	0.881	0.874	4.7	0	14	8300	0	0	no change

934  
 935

### CS3: Configuration-size optimisation - high renewable penetration

937 In this case study, we are looking for the optimum HRES with cheapest cost of energy for a  
 938 site with similar load and renewable resource profiles as in the previous case study. We also  
 939 add a constraint on the penetration, looking for systems with high renewable penetration ( $p \geq$   
 940 200%). Here, by adding a constraint on the penetration of the system we force the  
 941 optimisation process to reduce the share of the diesel in favour of the share of the renewables.  
 942

943 The obtained optimum configuration for this case study is in agreement with what we  
 944 expected by forcing the optimisation process toward a high penetration configuration. The

945 LCE is slightly more than that of CS2 explaining why diesel generator was kept in the  
946 configuration through the optimisation process of CS2.

947

948 Figure 5 shows the GA search history for this case study. Compared to the GA search  
949 histories of CS1 and CS2, a slower convergence rate is observed. This is due to searching  
950 within a more constrained design space and using the method of rejection of infeasible  
951 solutions.

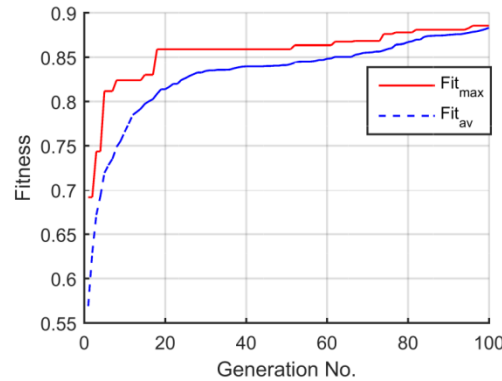


Figure 5-GA search history for CS3

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#### 955 **CS4: Size optimisation - fixed configuration**

956 In CS4, the configuration of wind-fuel cell/electrolyser is pre-planned by setting the upper  
957 and lower bounds for the size of the other components to zero. This is obviously a non-  
958 optimum configuration due to the site's low wind profile and the relatively high LCE of fuel  
959 cell/electrolyser compared to other renewable systems. The results show how expensive is  
960 this configuration in terms of LCE and TLSC for this site with cost parameters used in this  
961 paper, and in fact, explains why there is no fuel cell/electrolyser included in the optimum  
962 configurations found in the case studies 1 through 3.

963

964 The GA search history is shown in Figure 6. Compared to the search histories of CS1 through  
965 CS3, the search history for this case study has no major jumps in the maximum fitness in  
966 earlier generations. This was expected as in this case study the configuration is fixed and the  
967 major jumps in the earlier generations are due to changes in the configuration (as observed in  
968 Figures 2 and 5, and Tables 4 and 5).

969

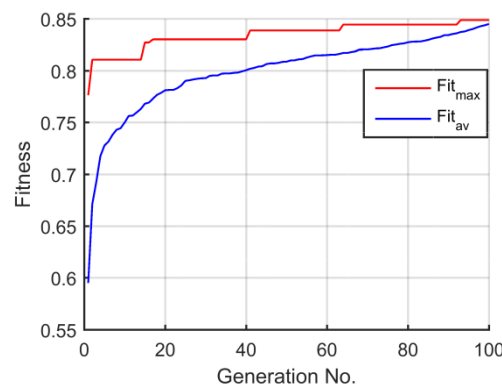


Figure 6-GA search history for CS4

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973 The results of this case study show the flexibility of the problem formulation in dealing with  
974 fixed configurations.



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### CS5: Configuration-size optimisation – retrofitting for saving and environment

In the retrofitting case of CS5 we already have a component with known size and would like to keep it as it is but add new components. In this case the upper and lower limits for diesel generator are set as the nominal size of the existing diesel generator ( $P_{D,nom} = 5kW$ ). Keeping the diesel generator in the configuration as an auxiliary unit reduces the overall cost of the system. However, the usage of the diesel is limited to an annual emission of  $CO_2 \leq 500 kg$ . In this case study, since the diesel generator is already in the system, the capital cost (initial cost + installation cost) of the diesel is excluded from the cost analysis.

The search history of Figure 7 shows a slower improvement rate in  $fit_{max}$  compared to those of CS1 and CS2. This is due to the presence of the tight constraint on the  $CO_2$  emission.

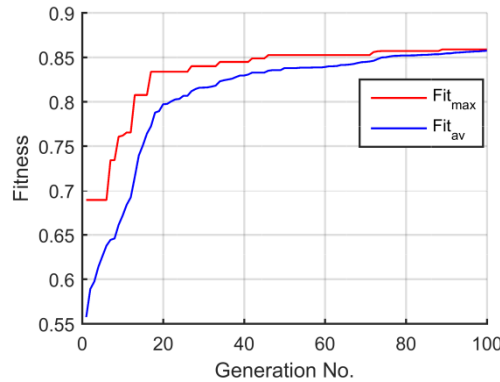


Figure 7-GA search history for CS5

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### CS6: Size optimisation – retrofitting for investment

In contrary to CS5, in which we retrofitted a given configuration without any constraint on the final configuration, here in CS6, the retrofitting aims at changing a given configuration to a fixed configuration (wind-PV-battery).

The existing 30kW wind turbine installed on Site 1 supplies only 71% of the 59.26 MW annual demand while the remaining 17.3 MW is supplied by the grid, which costs \$2,760 per year assuming a 16c/kWh flat rate price of electricity from grid. The purpose of retrofitting here is to eliminate the cost of buying electricity from the grid (reduce the unmet load  $U_t$  to zero) and to maximise the profit of selling electricity to the grid (assuming a flat rate feed-in tariff of 12c/kWh). The investment budget is limited to \$250,000 present value of total lifespan cost.

Similar to CS5, in this case study the capital cost of the wind turbine, as an existing component of the configuration, is excluded from the cost analysis. Unlike the first five case studies, in which infeasible solutions are rejected during the search process, here the constraint of  $TLSC \leq 250,000$  makes the design space highly constrained. Therefore, a penalisation strategy is adopted instead. For this problem, different penalty weights were tried to determine the values that put enough pressure on the search to enter the feasible domain before the end of the search. A penalty weight of  $\mu_i = 10$  for both constraints was found to be strong enough to lead to feasible solutions. The search history is shown in Figure 8. A low average fitness is observed at earlier generations, which is due to applying high penalties to infeasible solutions.

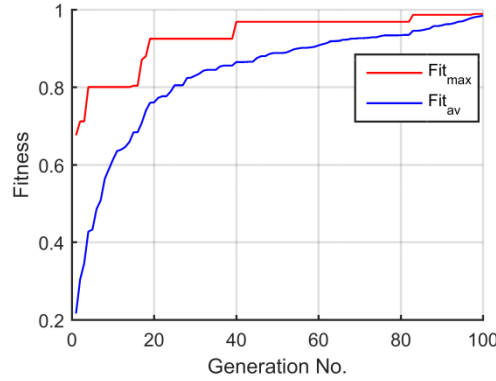


Figure 8-GA search history for CS6

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1018 With reference to Figures 2, and 4 through 8, one can observe that in all search histories a  
1019 good diversity (the distance between  $fit_{max}$  and  $fit_{av}$  curves) is observed for most of the  
1020 generations (up to about  $i_{gen} \sim 80$ ). However, towards the end of the search the diversity  
1021 starts to decrease rapidly. This is due to the dynamic nature of the designed mutation operator  
1022 at the size level (Equations 50 to 52). The mutation interval shrinks as  $i_{gen}$  increases. Hence,  
1023 mutation becomes less effective in exploration and more effective in exploitation. An  
1024 exploitative mutation together with a crossover, both based on a fitness-based parent  
1025 selection, lead to a rapid increase in the average fitness. This behaviour is an intended  
1026 behaviour with the aim of refining the optimum solution towards the end of the search.

1027

1028 The results of case studies 5 and 6 prove the flexibility of the formulation in delivering  
1029 different cases of retrofitting of existing power systems with adding new constraints or  
1030 objectives.

1031

### 1032 **CS7: Multiobjective integrated configuration-size optimisation**

1033 As mentioned earlier, conducting multiobjective optimisations is the best way to get an  
1034 insight to the interaction between the design qualities that we are interested in and to select a  
1035 suitable solution via an informed decision making process. This case study is a multiobjective  
1036 version of CS1, in which  $U_t$  is treated as an objective instead of a hard constraint. Figure 9  
1037 and Table 6 show the nondominated solutions for this case study. It should be noted that the  
1038 constraint  $p \geq 60\%$  here is an arbitrary constraint. The purpose of applying arbitrary  
1039 constraints is to put a focus on part of the design space that we are interested in. For instance,  
1040 without the constraint  $p \geq 60\%$  we would see hundreds of nondominated solutions  
1041 representing very small systems without any practical use due to very high unmet load.

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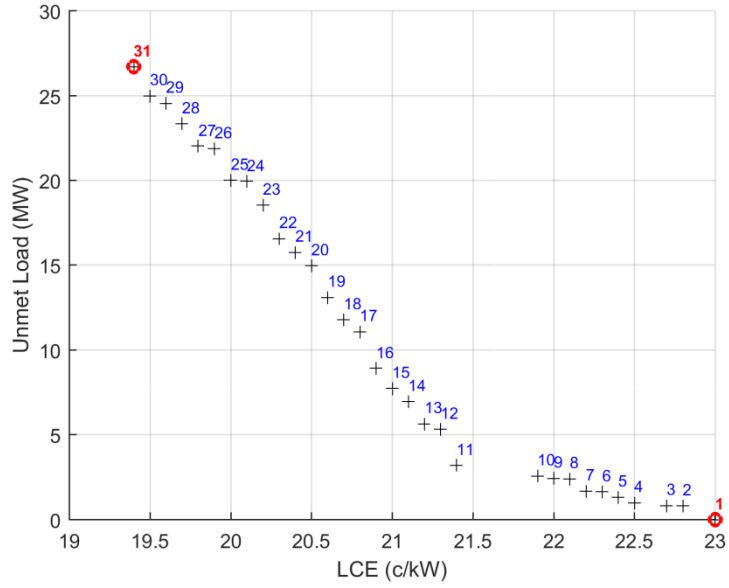


Figure 9-Pareto front of CS7

Table 6-Nondominated solutions of CS7

Sol #	$R_{WT}$ ( $m^2$ )	$A_{PV}$ ( $m^2$ )	$n_B$	$P_{D,nom}$ (W)	$P_{FC,nom}$ (W)	$P_{EL,nom}$ (W)	TLSC (1000\$)	LCE (c/kWh)	$U_t$ (MW)	$p$ (%)	$CO_2$ (kg)
1	0	296	232	0	0	0	186	23.0	0.00	139	0
2	0	289	224	0	0	0	181	22.8	0.83	135	0
3	0	281	232	0	0	0	180	22.7	0.85	132	0
4	0	277	232	0	0	0	178	22.5	1.01	130	0
5	0	271	234	0	0	0	177	22.4	1.34	127	0
6	0	264	238	0	0	0	175	22.3	1.67	124	0
7	0	263	234	0	0	0	174	22.2	1.71	123	0
8	0	254	236	0	0	0	171	22.1	2.43	119	0
9	0	259	224	0	0	0	170	22	2.46	121	0
10	0	256	224	0	0	0	169	21.9	2.60	120	0
11	0	231	214	700	0	0	163	21.4	3.22	108	677
12	0	235	202	0	0	0	156	21.3	5.35	110	0
13	0	232	200	0	0	0	154	21.2	5.66	109	0
14	0	224	196	0	0	0	150	21.1	6.98	105	0
15	0	216	198	0	0	0	147	21	7.74	101	0
16	0	208	194	0	0	0	143	20.9	8.92	97	0
17	0	206	166	0	0	0	136	20.8	11.11	97	0
18	0	194	176	0	0	0	134	20.7	11.81	91	0
19	0	189	166	0	0	0	129	20.6	13.13	89	0
20	0	183	150	0	0	0	123	20.5	14.99	86	0
21	0	183	138	0	0	0	121	20.4	15.78	86	0
22	0	173	144	0	0	0	118	20.3	16.59	81	0
23	0	164	132	0	0	0	112	20.2	18.57	77	0
24	0	157	126	0	0	0	107	20.1	19.95	74	0
25	0	158	122	0	0	0	107	20	20.01	74	0
26	0	148	116	0	0	0	101	19.9	21.88	69	0
27	0	148	112	0	0	0	100	19.8	22.03	69	0
28	0	143	102	0	0	0	96	19.7	23.36	67	0
29	0	136	100	0	0	0	93	19.6	24.56	64	0
30	0	135	94	0	0	0	91	19.5	25.00	63	0
31	0	127	86	0	0	0	86	19.4	26.71	60	0

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1049 Solutions #1 and #31 are the extreme (far-end/boundary) solutions. In a two-dimensional  
1050 Pareto front, an extreme solution is the best in terms of one objective and the worst in terms  
1051 of the other objective. Solution #1  $\{U_t, LCE\} = \{0, 23\}$  is the best solution in terms of  $U_t$  and  
1052 the worst in terms of  $LCE$ . Once the Pareto front is obtained, we can conduct a trade-off study  
1053 towards selecting a suitable solution. For instance, comparing Solution #12 with the most  
1054 reliable solution (Solution #1), one can see that \$30,000, one sixth of the  $TLSC$ , can be saved,  
1055 if we can tolerate an unmet load of  $U_t = 5.35 MW$ , which is only 9% of the total annual  
1056 demand load of 59.26 MW.

1057

1058 The performance of a multiobjective optimisation method in producing a good approximation  
1059 of the Pareto front can be assessed by examining two factors: (i) its capability in producing a  
1060 uniformly populated front and (ii) the (global) optimality of the extreme solutions on that  
1061 front. With reference to Figure 9, it is evident that the developed NSGA-II meets the first  
1062 assessment criterion as it produces a fairly uniform and well populated Pareto front. To test  
1063 the NSGA-II performance in terms of the second assessment criterion, we know that  
1064 theoretically, the extreme solutions on the actual Pareto front are the results of single  
1065 objective optimisation problems in which one design quality is considered as objective and  
1066 the other as constraint. All we need to do is to run two single objective optimisation problems  
1067 and compare their results with the extreme solutions on the front. The accuracy of the  
1068 extreme Solution #1 can be examined by solving the single objective problem:  
1069 *minimise*  $LCE$  subject to  $U_t = 0$ . We already have solved this problem in CS1 and have  
1070 proved (using exhaustive search) that  $\vec{X}_{opt} = \{0, 296, 232, 0, 0, 0\}$  (see Table 2) is the global  
1071 optimum, which is identical to the Solution #1 obtained by NSGA-II. This shows that the  
1072 NSGA-II finds accurate extreme solutions and therefore passes the second assessment  
1073 criterion.

1074

### 1075 **CS8: Multiobjective configuration-size optimisation - low renewables sites**

1076 Nowadays most of renewable technologies have competitive prices compared to fuel-based  
1077 power systems. As a result of this, in sites with medium to high renewable resources a  
1078 configuration without diesel generator is probably the optimum solution in terms of the cost  
1079 of produced energy (for instance, see CS1). For sites with low renewable resources we expect  
1080 to see diesel in the optimum system configuration. In such cases the environmental impact  
1081 due to the  $CO_2$  emission needs to be considered alongside the cost and reliability of the  
1082 system. Site 3 is a low renewable site. Here, we are dealing with three conflicting objectives:  
1083 a cost related ( $LCE$ ), a reliability related ( $U_t$ ), and an environment related ( $CO_2$ ).

1084

1085 The multiobjective optimisation leads to 81 nondominated solutions. The results of the  
1086 optimisation are shown in Figure 10. The extreme solutions are identified with coloured  
1087 circles. In Figure 10 the solution identified with red circle is the best in terms of  $LCE$   
1088 ( $LCE = 29.0 c/kWh$ ), those identified with blue circles are the best in terms of  $U_t$  ( $U_t = 0$ ),  
1089 and the green ones are those solutions without diesel generator in the configuration and  
1090 therefore the best ones in terms of  $CO_2$  emission ( $CO_2 = 0$ ).

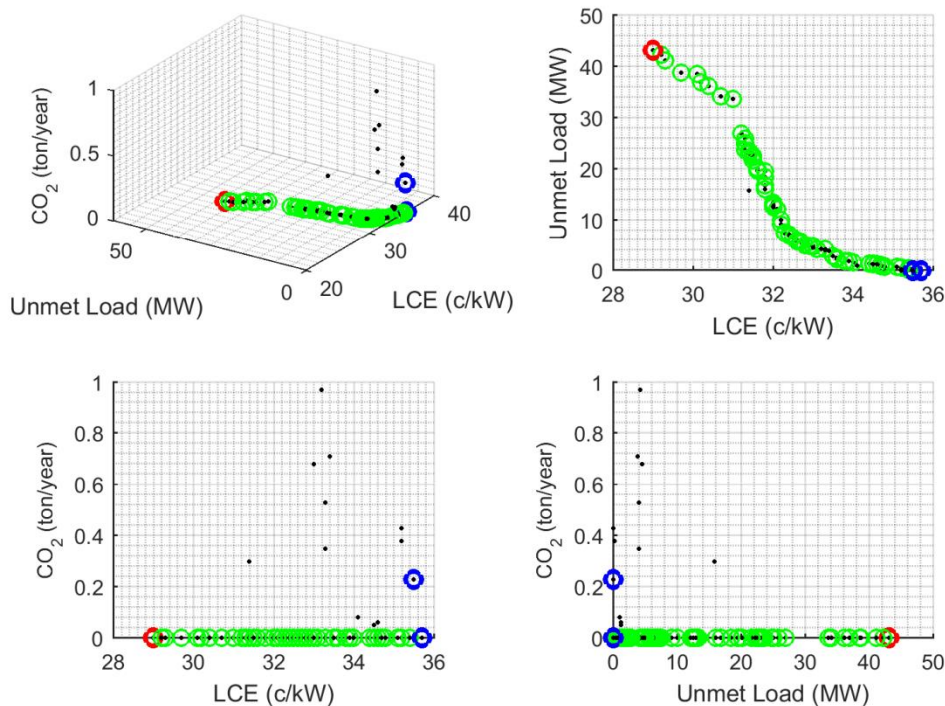


Figure 10-3D Pareto front (top left) and pair-wise fronts of CS8

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#### 4.2 Summary of the Results of Case Studies

The results of the case studies prove that the new integrated size-configuration problem formulation has the following features:

- 1099 • Flexibility: It addition to starting from a generic configuration (which includes all  
1100 possible components) and leading to the best configuration for a given site (CS1, CS2,  
1101 CS3, CS7 and CS8), it can also deal with pre-defined (fixed) configurations (CS4),  
1102 retrofitting of an existing system to a generic configuration (CS5), and retrofitting of an  
1103 existing system to a fixed configuration (CS6).
- 1104 • The problem formulation treats all design variables as continuous variables, allowing the  
1105 manufacturer and provider companies to develop bespoke components for a site to reduce  
1106 the cost/increase the reliability (e.g. by manufacturing a wind turbine with the required  
1107 rated power instead of using an existing wind turbine which is oversized or under-  
1108 designed for that site).
- 1109 • Versatility: It can be easily expanded to include more components than those considered  
1110 in this paper (wind turbine, PV panel, battery bank, fuel cell, electrolyser and diesel  
1111 generator), and to include design qualities (objective and constraints) beyond those  
1112 considered in this paper.

1113

The results also show that the GA reproduction operators which are developed specifically  
1115 for the problem at hand have the following characteristics:

- 1116 • Excellent exploitatory functionality at size level: The design space is very vast at the size  
1117 level. There are 7 design variables, of which 6 of them have a wide range (all but  $n_{WT}$ ).  
1118 For instance for the case studies 1, 2, 3, 7 and 8 the lower and upper limits for the design  
1119 variables are  $\vec{X}^l = \{1,0,0,0,0,0,0\}$  and  $\vec{X}^u = \{1,22, 460, 1030, 100000, 8800, 55000\}$   
1120 with physical increments of  $\{0,0.1, 1, 1, 100, 100, 100\}$ . In order to ensure that the GA

1121 and NSGA-II work efficiently and find optimum/nondominated solutions even with a  
1122 small population size (in this study 20) and within a small generation number (here 100),  
1123 the exploitation at the size level must be very efficient. This has been achieved by using a  
1124 combination of a weighted arithmetic crossover and a dynamic mutation. It was shown  
1125 that this combination provides excellent exploitation at size level, leading to solutions  
1126 which match the results of a refined exhaustive search (optimum solution of CS1, and  
1127 extreme solution of CS7).

- 1128 • Robust exploratory functionality at size level: The exploratory function is of prime  
1129 importance for problems with piecewise quality space (see separate surfaces for LCE in  
1130 Figure 3). A search algorithm with poor exploration capability is more likely to remain on  
1131 one of the surfaces and therefore to get trapped in a local optima. The search histories of  
1132 CS1 through CS6, show that the population diversity remains high for most of the  
1133 generations (indicating a good exploratory functionality), until towards the end of the  
1134 search where, as intended, the exploitative functionality of the mutation takes over (fast  
1135 reduction in the diversity).
- 1136 • Efficient exploratory functionality at configuration level: Compared to the design space at  
1137 size level, the design space at configuration level is discrete and limited. Therefore,  
1138 exploitation is not applicable at configuration level. On the other hand, an efficient  
1139 exploration at configuration level is very important to have an overall robust optimisation  
1140 algorithm. An efficient exploration at configuration level allows to identify those  
1141 configurations which are more likely to be optimum at earlier generations and then, if  
1142 necessary, to switch between potential optimum configurations as the design space at size  
1143 level is explored and exploited. Removing components from the configuration  
1144 (downward change) and adding components to a configuration (upward change) are  
1145 delivered by a combination of crossover and mutation. The arithmetic crossover can lead  
1146 to upward change in configuration (but not downward). Mutation can change the  
1147 configuration both downward and upward. Tables 4 and 5 of case studies CS1 and CS2  
1148 show the evolution of the best configuration according to what we expect to see from the  
1149 planned exploration mechanism at configuration level.
- 1150 • Capability of handling highly constrained problems: In single objective optimisation  
1151 formulation, the constraints can be defined on any of the design qualities  
1152 ( $TLSC$ ,  $LCE$ ,  $U_t$ ,  $CO_2$ ,  $p$ ,  $feed$ , and  $rofit$ ). This may lead to highly constrained problems,  
1153 in which penalising infeasible solutions is the only way of evaluating solutions (e.g. as in  
1154 CS6). Design qualities have different units and different order of magnitudes. The penalty  
1155 function defined for handling infeasible solutions in highly constrained problems is based  
1156 on normalised deviations and therefore needs minimal tuning.
- 1157 • Capability of producing a uniformly populated front and finding the extreme solutions  
1158 accurately as shown in CS7 and CS8 (for NSGA-II).

1159  
1160

## 1161 **5 Concluding Remarks**

1162 Sequential configuration-size optimisation, as the current practice, needs a complete size  
1163 optimisation for all potential configurations that can be made by using all potential  
1164 components in the system. This can be a cumbersome task when dealing with many potential  
1165 components. The reported integrated configuration-size optimisation formulation in this  
1166 paper allows finding the optimum configuration for a given site and the optimum size of each  
1167 component in that configuration by solving only one optimisation problem. This formulation  
1168 allows a search within the configuration and size domains simultaneously, hence exploring  
1169 the overall design space more rigorously, leading to superior solutions.

1170

1171 The integrated configuration-size optimisation formulation presented in this paper can be also  
1172 used for retrofitting of an existing system, which is another valuable capability of the  
1173 reported formulation when we design energy transition scenarios. Only an integrated size-  
1174 configuration formulation allows to incorporate all characteristics of an energy transition  
1175 scenario, such as adding new sources of energy and new storage systems to the existing  
1176 system, taking into account an increase in the demand load, and assigning targets to the  
1177 performance measures (e.g. a targeted reduction in emission).

1178  
1179 The integrated configuration-size optimisation formulation requires to be accompanied by a  
1180 robust search algorithm, which is capable of exploration and exploitation of the design space  
1181 at configuration and size levels. The developed GA and NSGA-II are proved to be robust in  
1182 exploration and exploitation of the design space at both configuration and size level. These  
1183 optimisers can be easily adapted by other researchers and expanded to include more design  
1184 variables and design qualities.

1185  
1186 The generality of the problem formulation and the robustness of the developed optimisers  
1187 allow for the expansion of the current work beyond power production in future studies, for  
1188 instance, by integrating heating and thermal storage components and optimal design of  
1189 combined heat/power systems.

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## 1361 Appendix

1362 Table A.1-The remaining power and cost modelling parameters

Parameter (unit)	Value
Margin of safety (storage sizing) $MOS(-)$	0.2

Autonomy period (battery sizing) $T_{a,B}$ (day)	1
Autonomy period (H2 tank sizing) $T_{a,H_2}$ (day)	0.5
Air density $\rho$ ( $kg/m^3$ )	1.225
Overall wind turbine mechanical and electrical efficiency $\eta_{EM}$ (%)	90
Site surface roughness length $z_0$ (m)	0.03
Minimum blade tip-ground clearance $h_c$ (m)	8
Wind turbine hub elevation for pre-sizing calculations $h_{hub,0}$ (m)	12
PV panel efficiency $\eta_{PV}$ (%)	14
Battery nominal capacity per unit battery $c_B$ (Ah)	40
Battery bank voltage $V_B$ (V)	24
Battery maximum SOC, $SOC_{max}$ (-)	1
Battery minimum allowable SOC, $SOC_{min}$ (-)	0.5
Battery efficiency in charge $\eta_{B,c}$ (%)	90
Battery efficiency in discharge $\eta_{B,d}$ (%)	95
Battery self-discharge rate, $\delta$ (%)	0.2
Fuel cell efficiency $\eta_{FC}$ (%)	47
Electrolyser efficiency $\eta_{EL}$ (%)	74
Diesel generator fuel cost $C_{fuel}$ (\$/l)	1
Emission per litre of diesel $CO_2$ (kg)	2.68
System nominal life $N_S$ (years)	20
Real discount rate $d$ (%)	4

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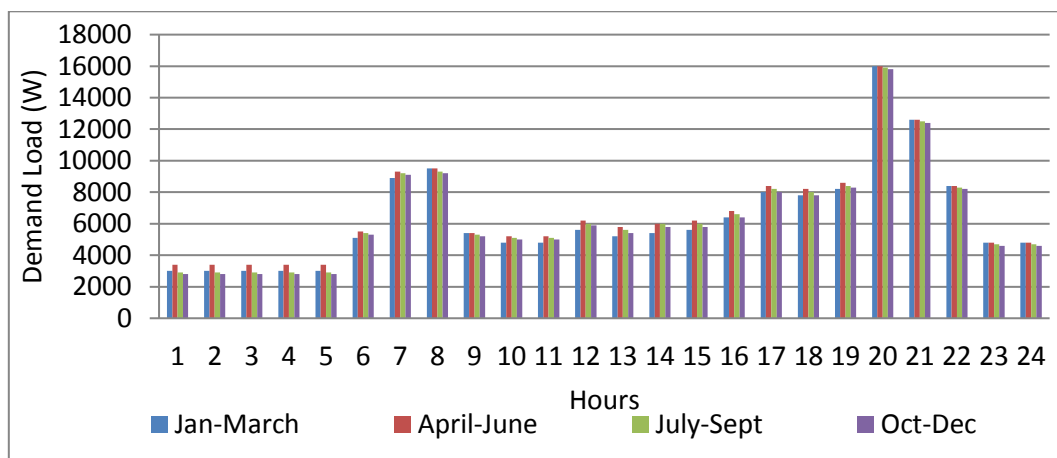


Figure A1-Hourly averaged demand load for four seasonal typical days

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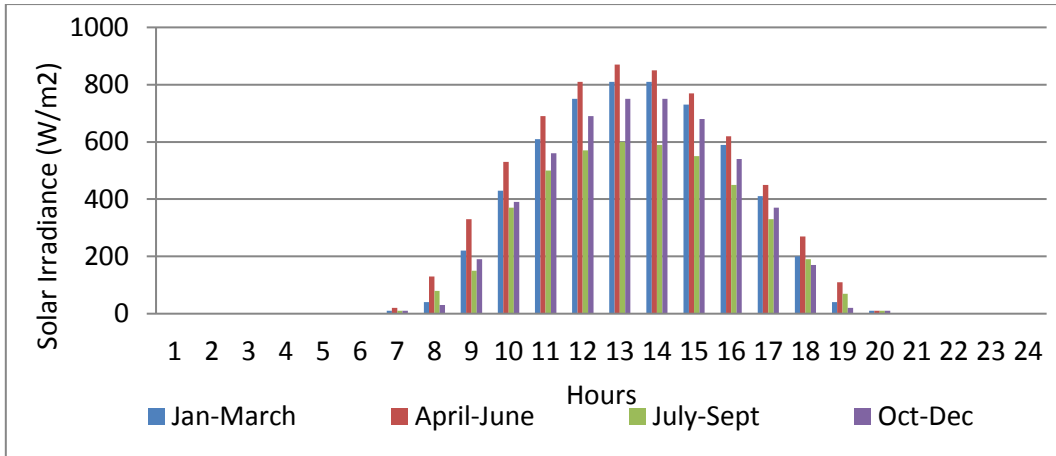


Figure A2-Hourly averaged solar irradiance for four seasonal typical days

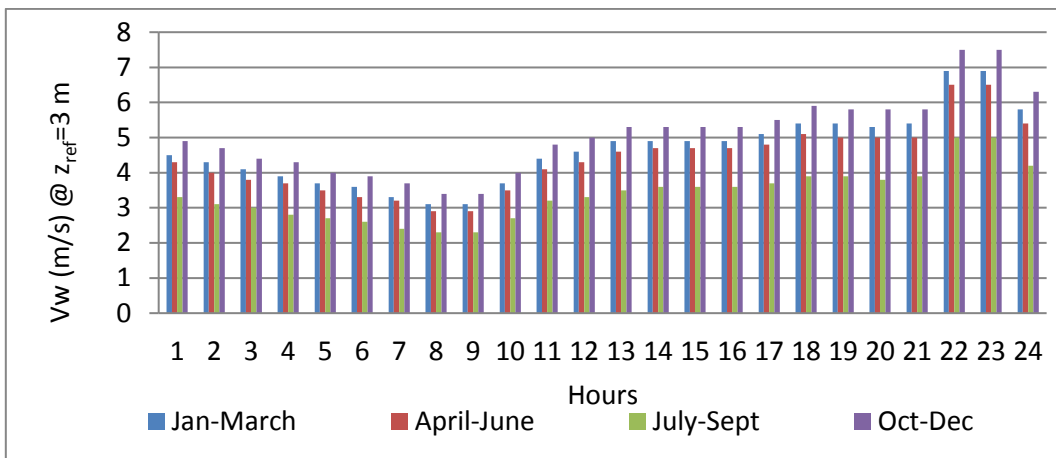


Figure A3-Hourly averaged wind speed for four seasonal typical days at a reference height of  $z_{ref} = 3m$

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